



**HERMES**

# Nano-satellites for high energy astrophysics and fundamental physics research

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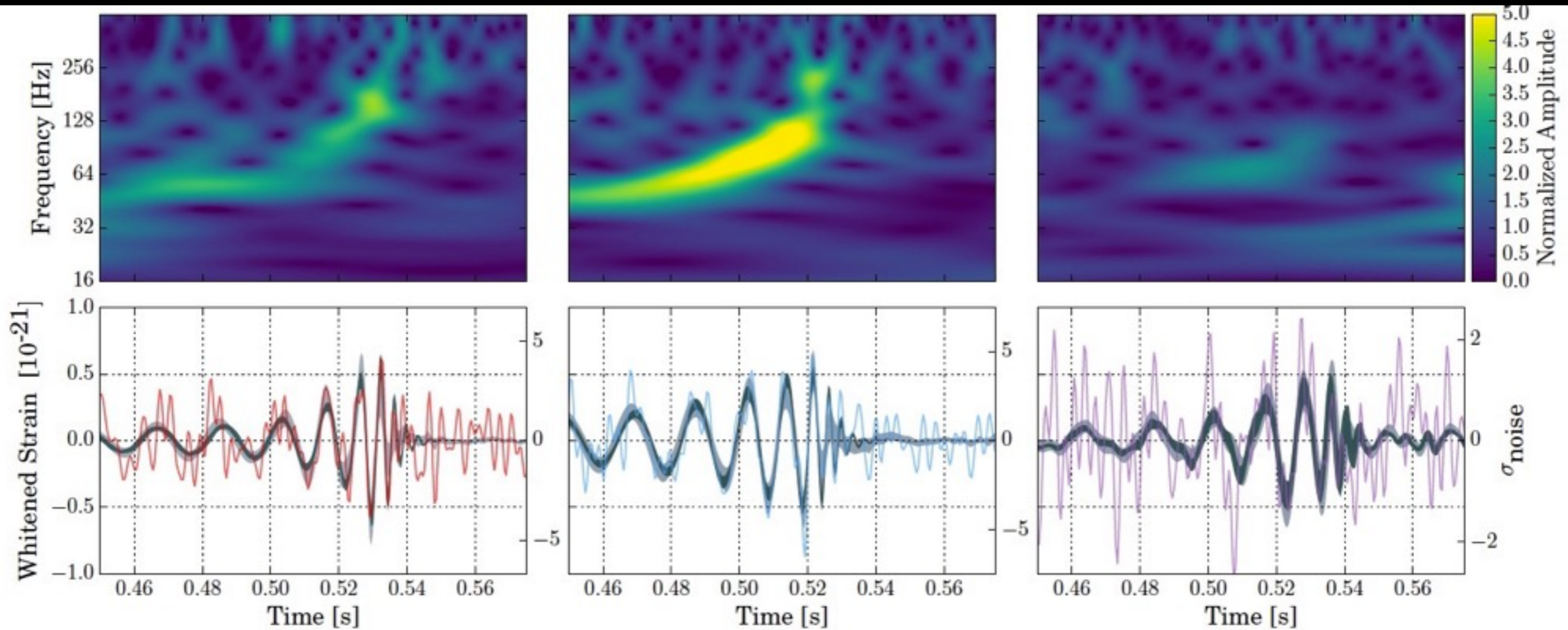
on behalf of the HERMES-TP and HERMES-SP collaborations



# Two revolutions

Multimessenger astrophysics

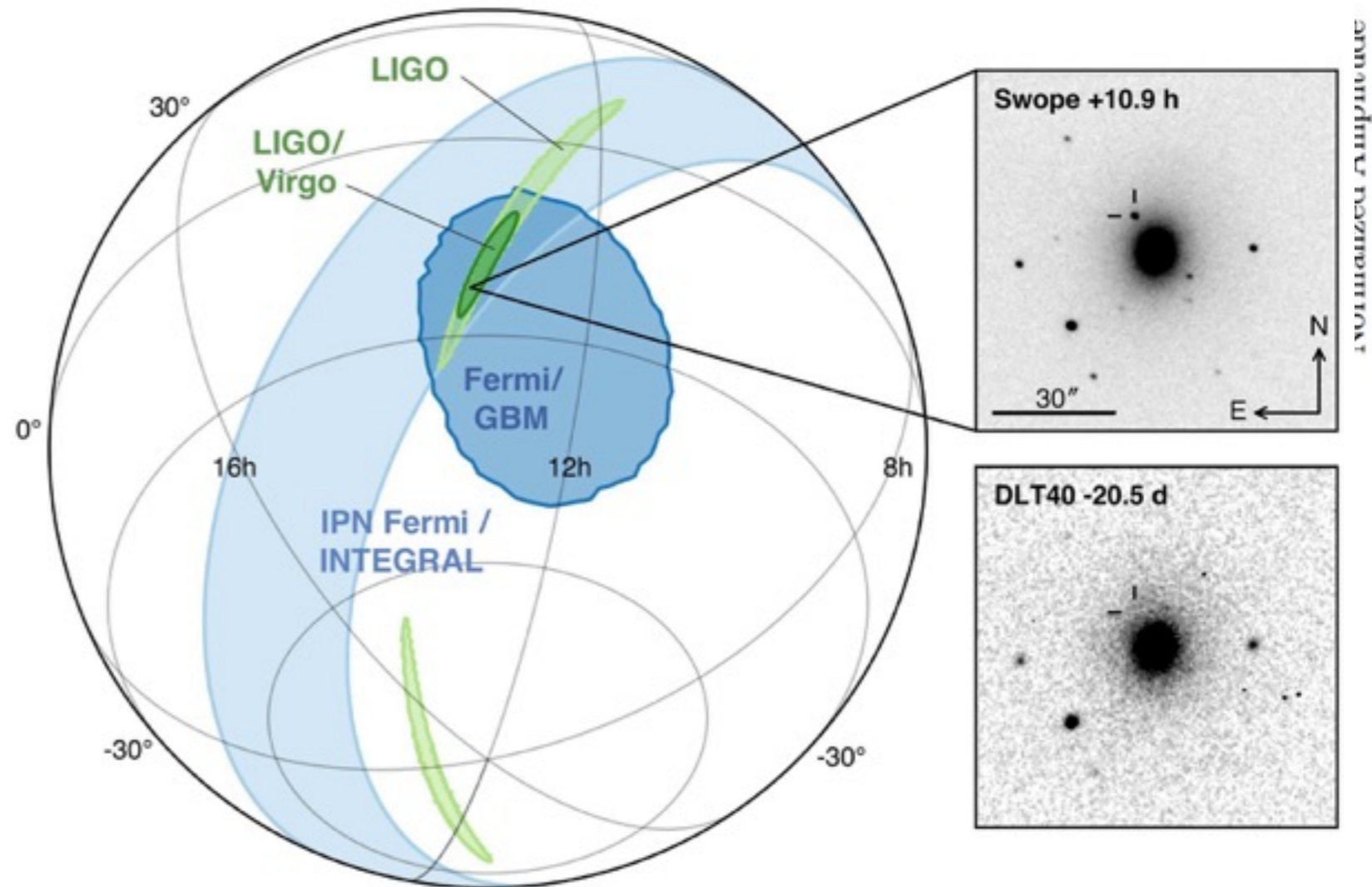
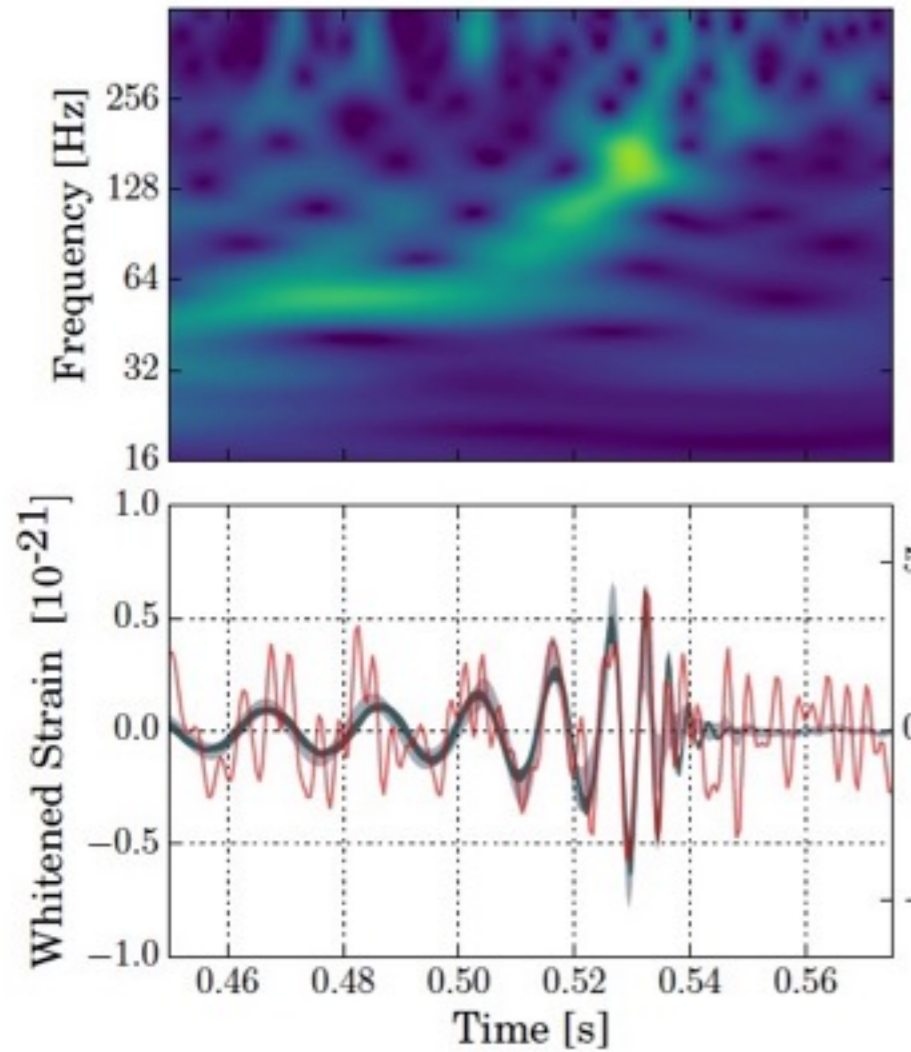
GW170814



# Two revolutions

Multimessenger astrophysics

GW170817



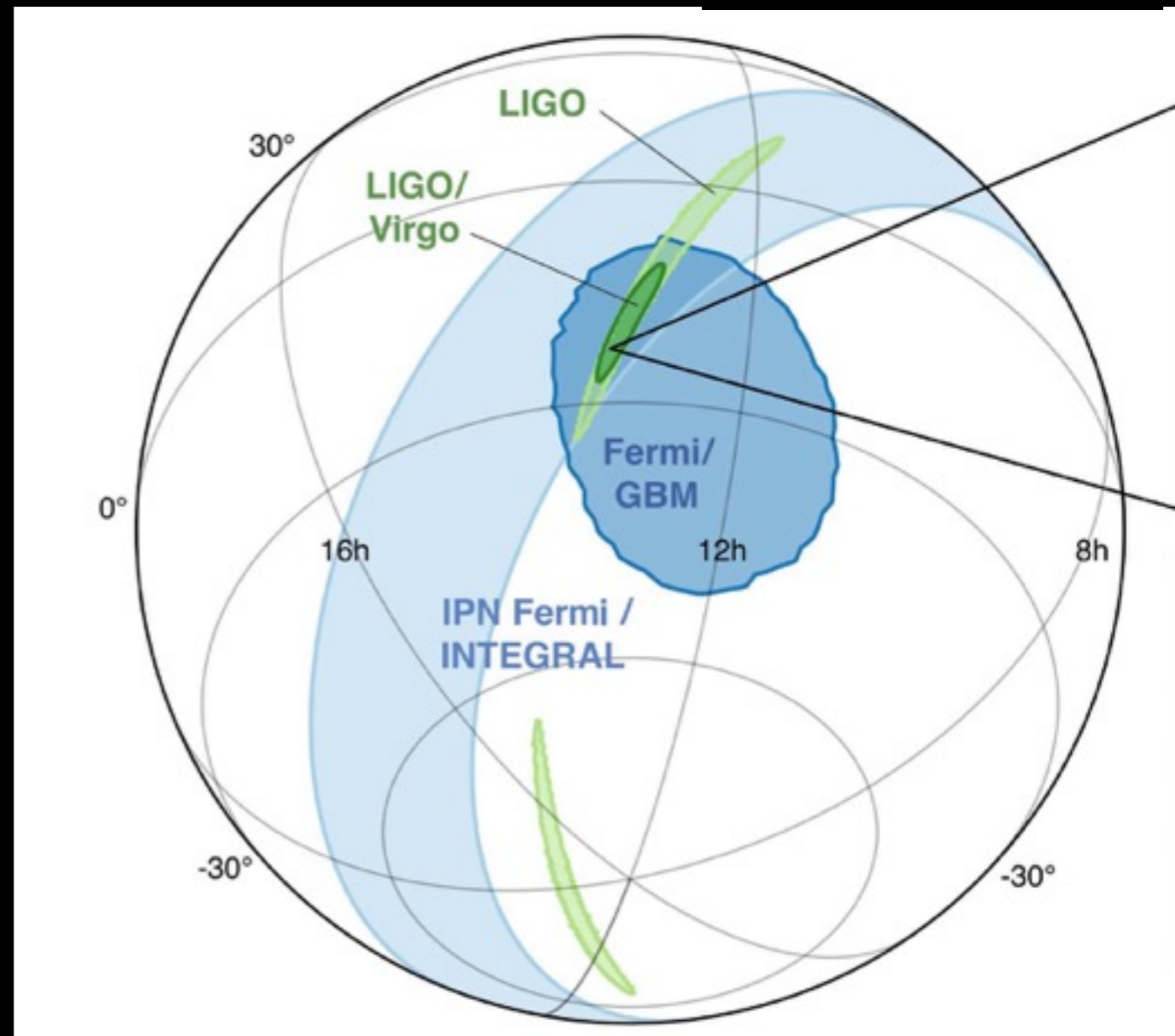
# The multimessenger revolution

Advanced Ligo/Virgo provide position with accuracy  
~ tens deg

NS-NS and BH-NS  
coalescence:  
100-200 Mpc horizon  
GRB, cocoon, kilonova..

BH-BH coalescence:  
>Gpc horizon  
no expected EM counterpart  
(even more exciting if one is  
found...)

GW170817



# The multimessenger revolution

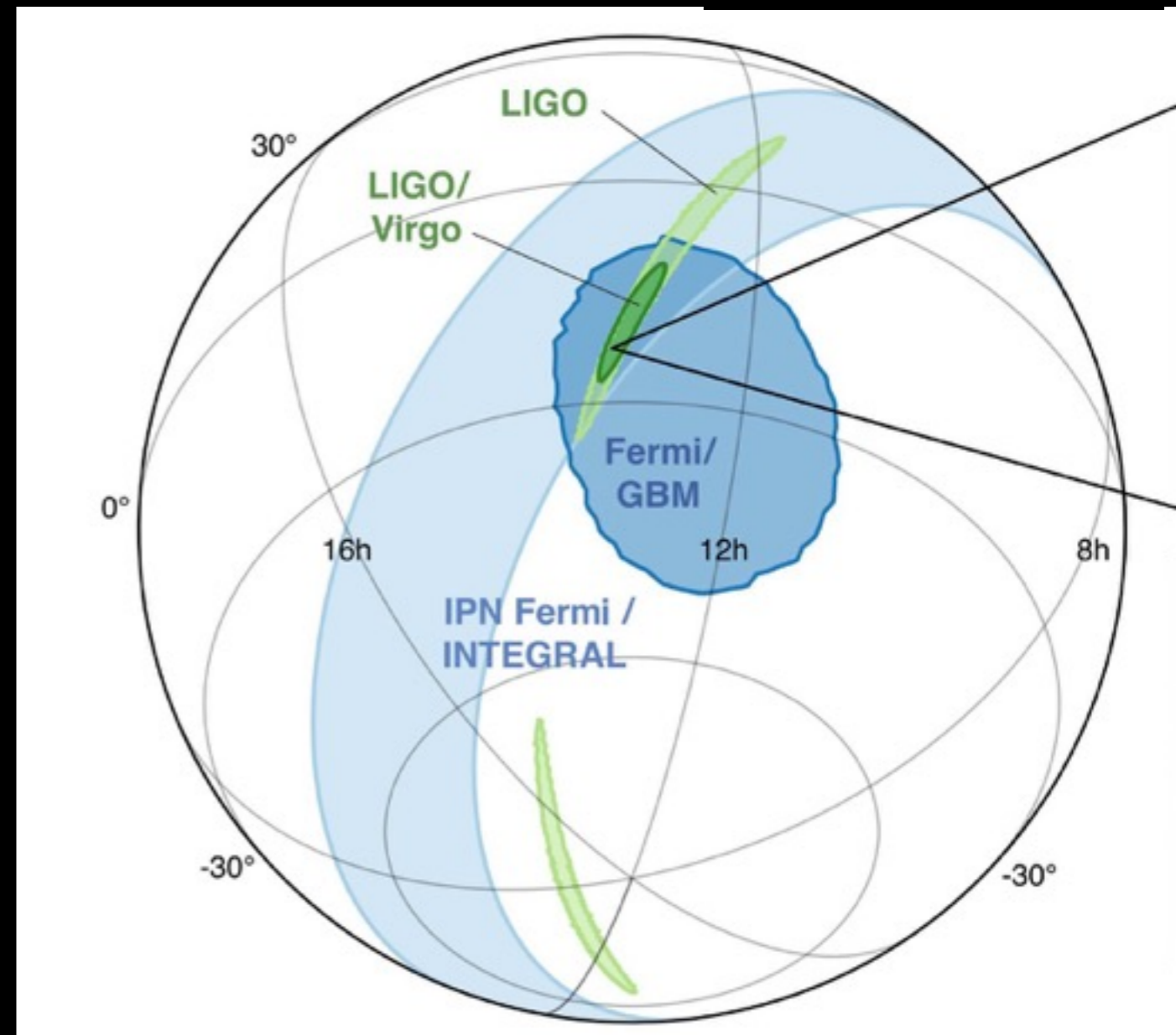
Large volumes difficult to survey at optical  $\lambda$ .

Tens/hundreds/thousands optical transients.

Best strategy:

~ all sky prompt search for transients at high energies. Negligible probability to find an uncorrelated HEA transient at the time of GWE

GW170817



# The multimessenger revolution

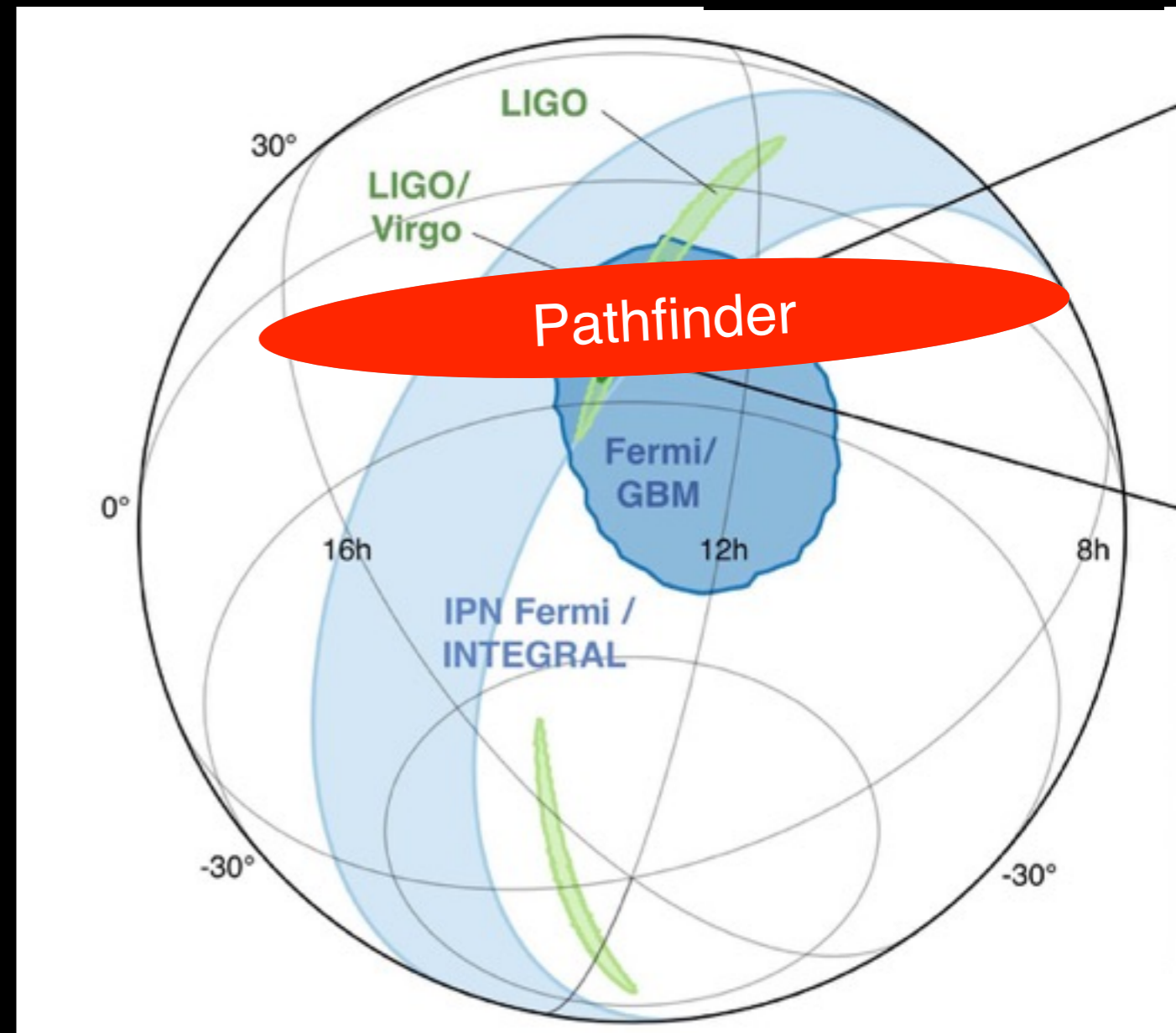
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GW170817



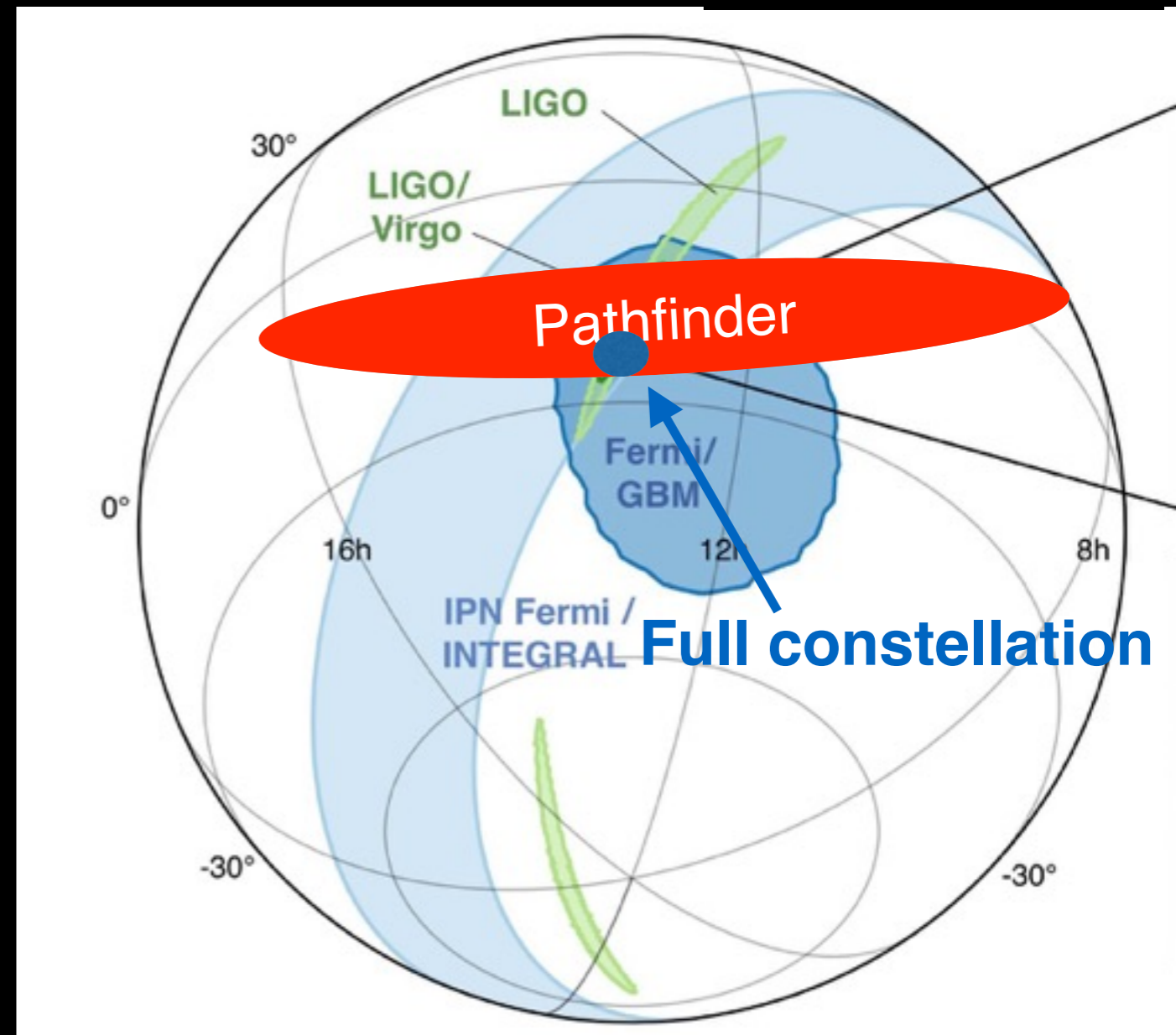
# The multimessenger revolution

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GW170817



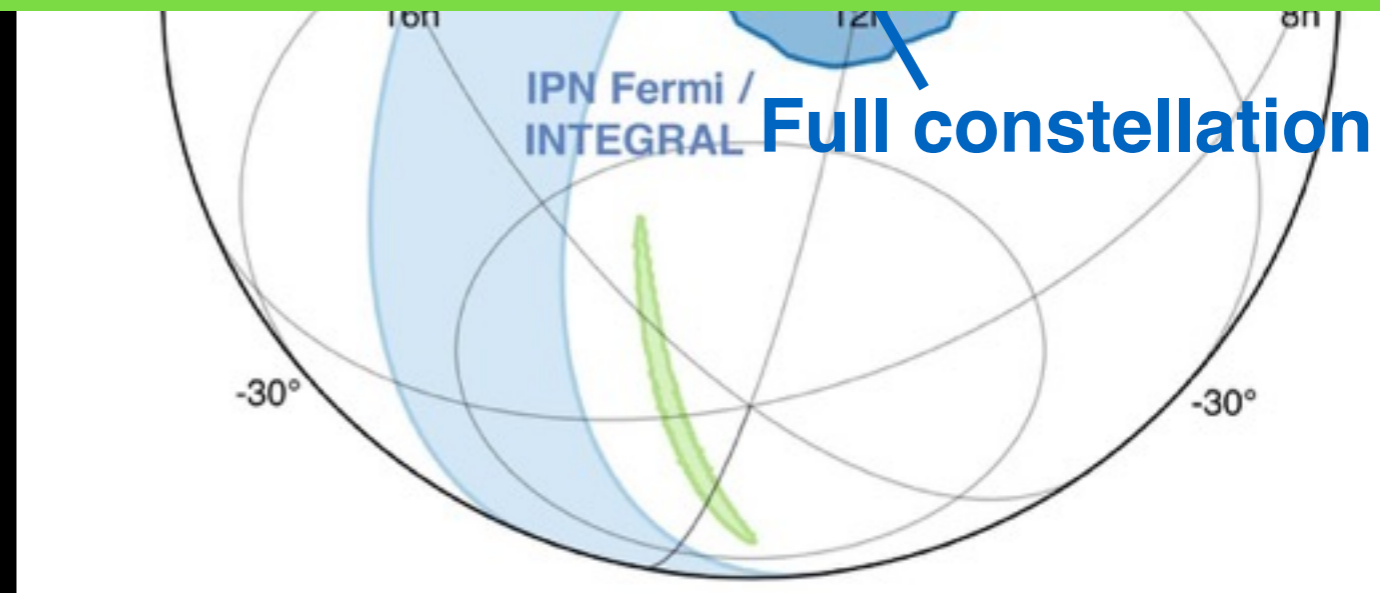
# The multimessenger revolution

**Current facilities, Swift, INTEGRAL, FERMI, AGILE, are aging:**

**A sensitive X-ray all sky monitor during the 20'**

Best strategy:

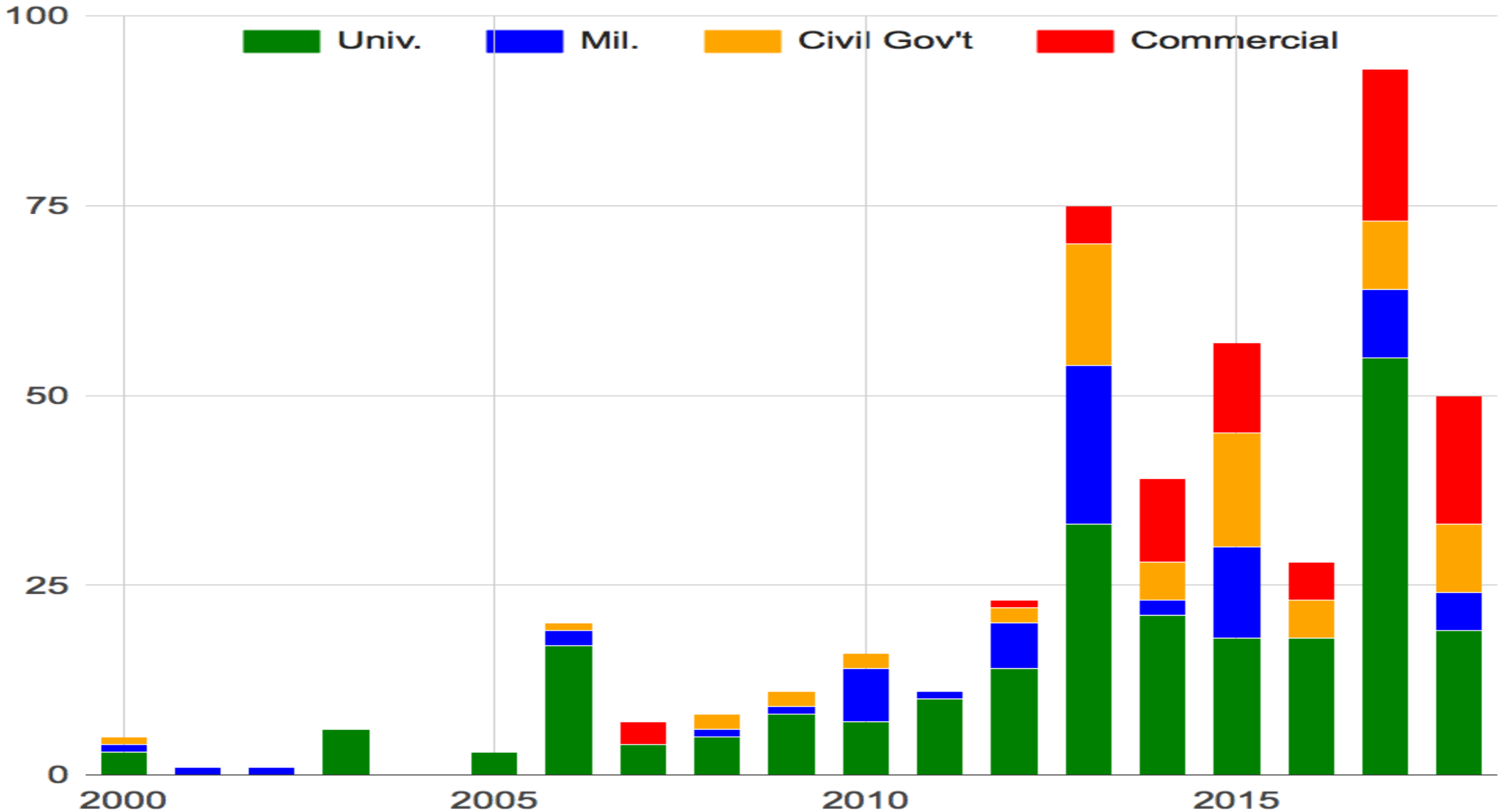
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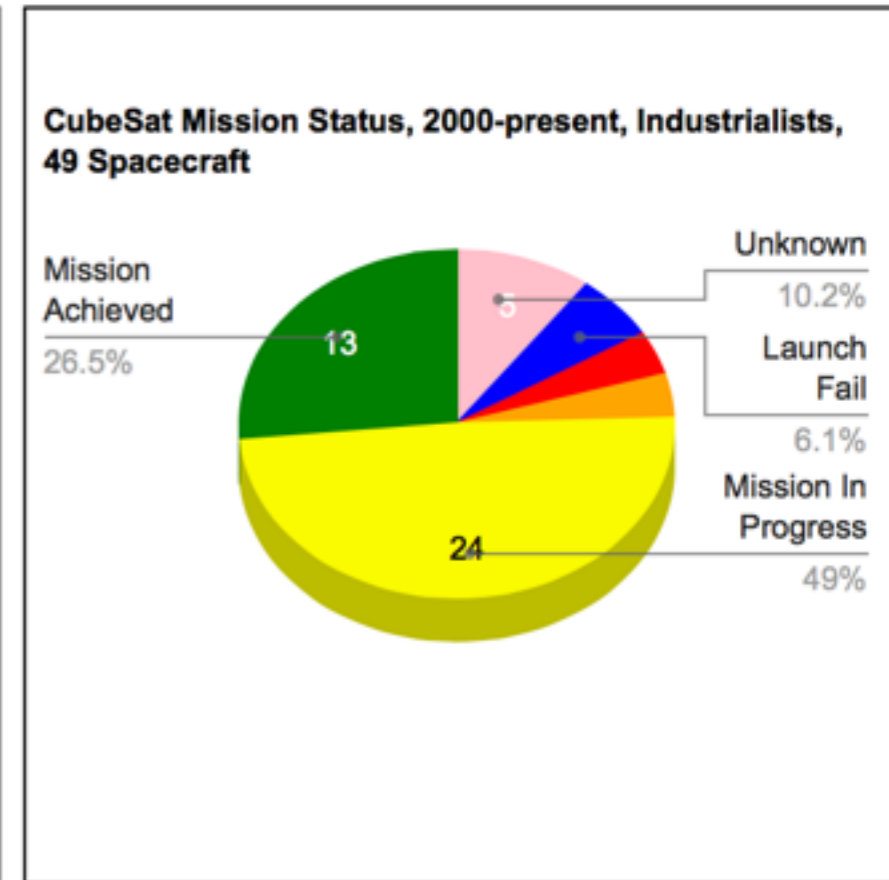
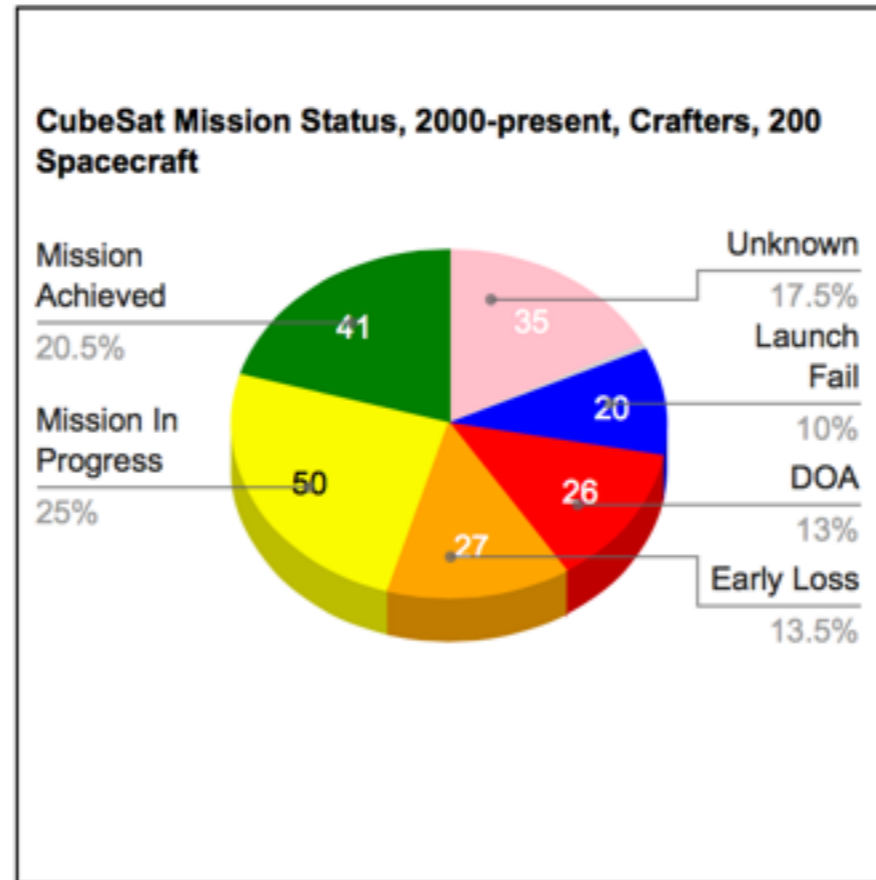
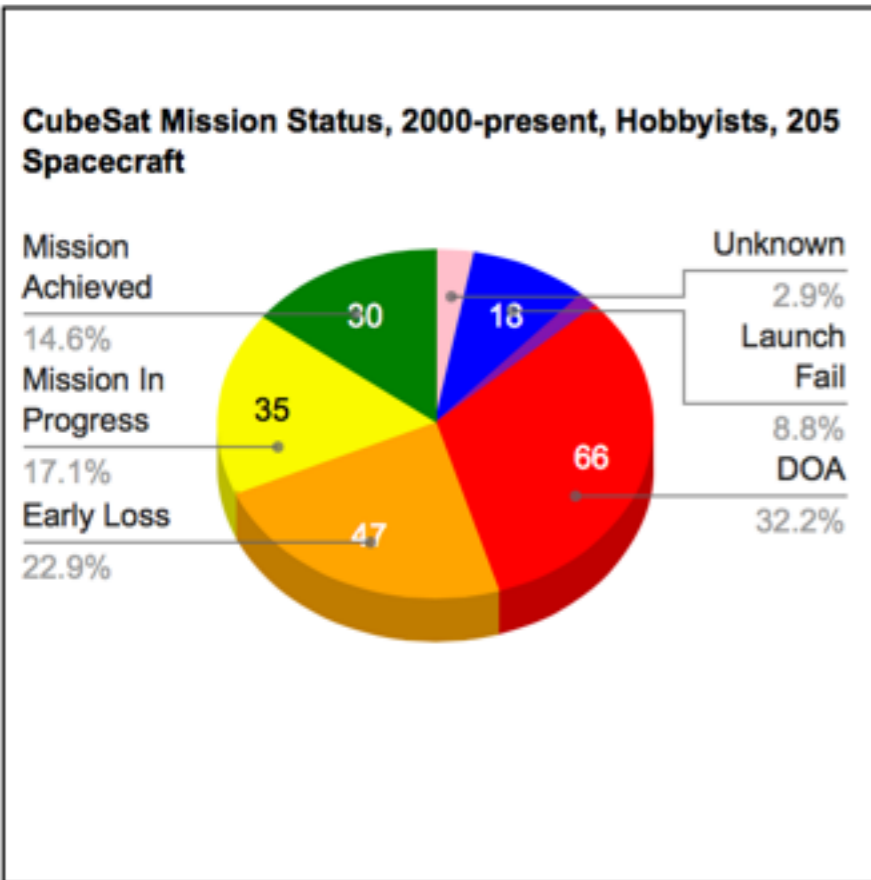
# Space 4.0

CubeSats by Mission Type (2000-present,



[Chart created on Wed Nov 14 2018 using data from M. Swartwout]

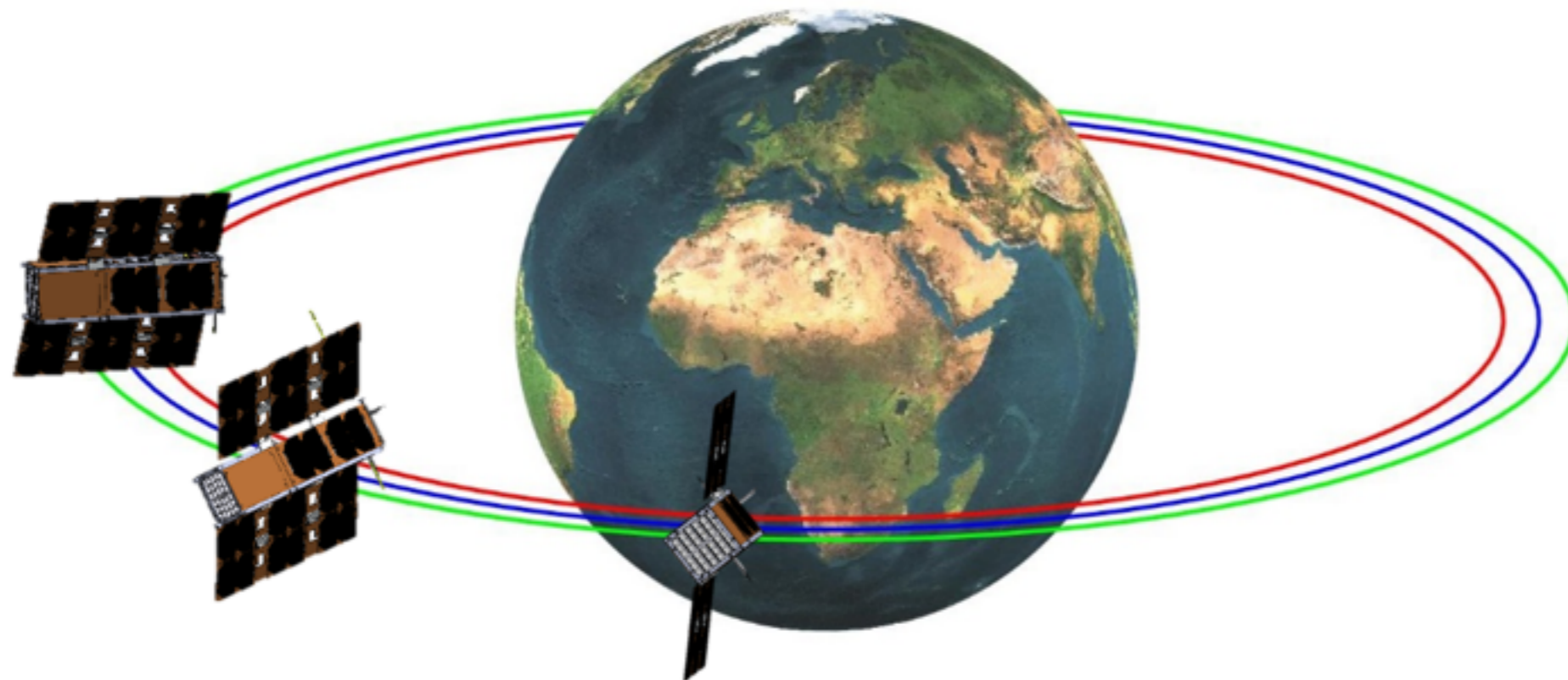
# Space 4.0





HERMES

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To Sun

# Mission concept

Disruptive technologies: cheap, underperforming, but producing high impact. Distributed instrument, tens/hundreds of simple units

## **HERMES constellation of cubesat**

2016: ASI funds for detector R&D

2018: MIUR funds for pathfinder

(Progetti premiali 2015)

2018 H2020 Space-SCI-20 project

2019 ASI internal funds



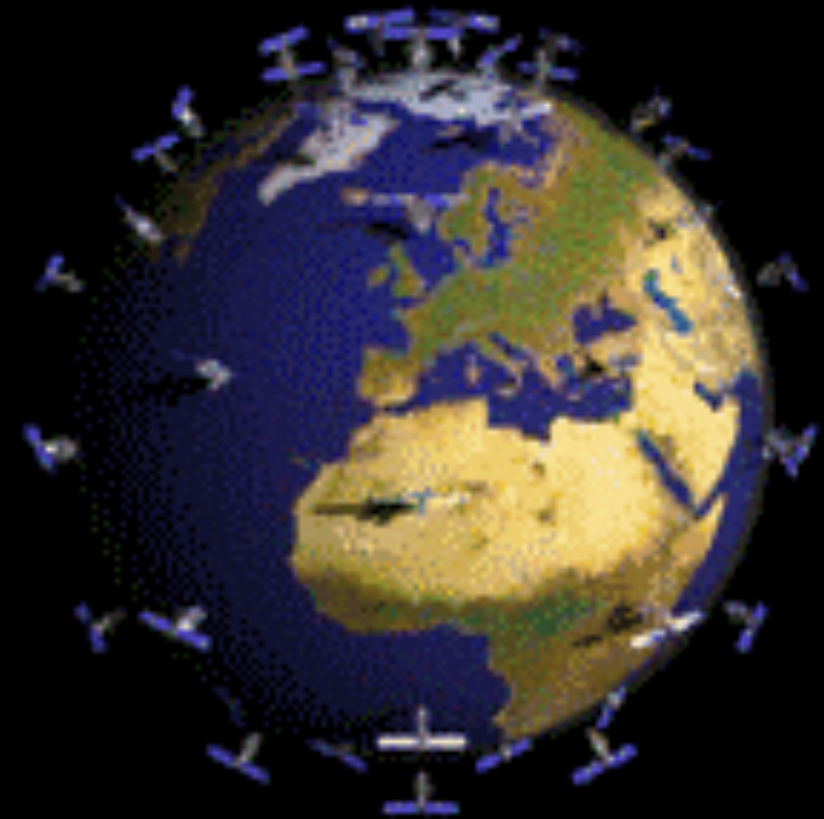
# Why HERMES now

## **Breakthrough scientific case:**

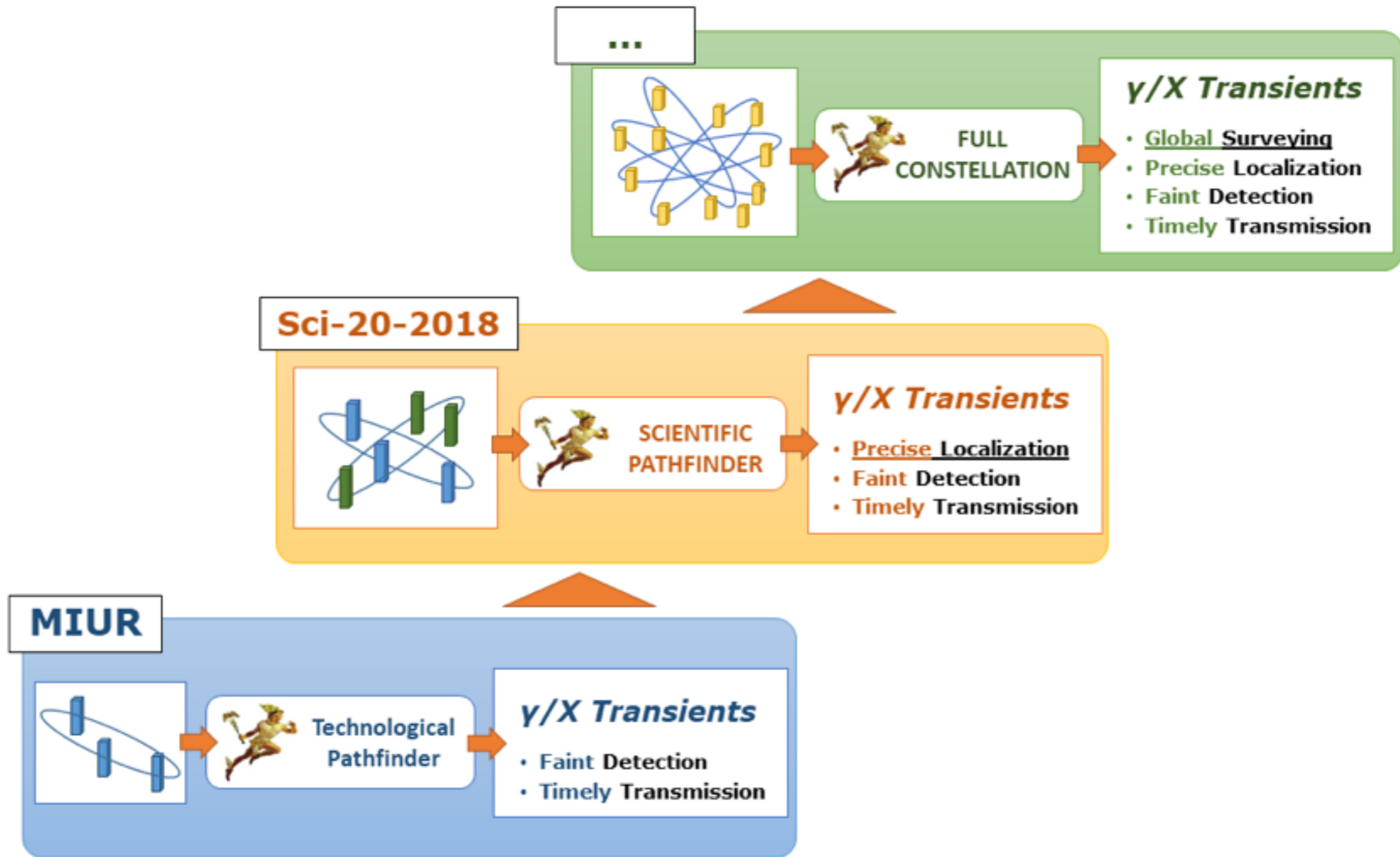
- EM of GWE

## **Modularity:**

- Avoid single point failures, improve hardware
- Pathfinder



# Why HERMES now



# Why **HERMES** now

## **Breakthrough scientific case:**

- EM of GWE

## **Modularity:**

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## **Open $\mu$ sec - msec window:**

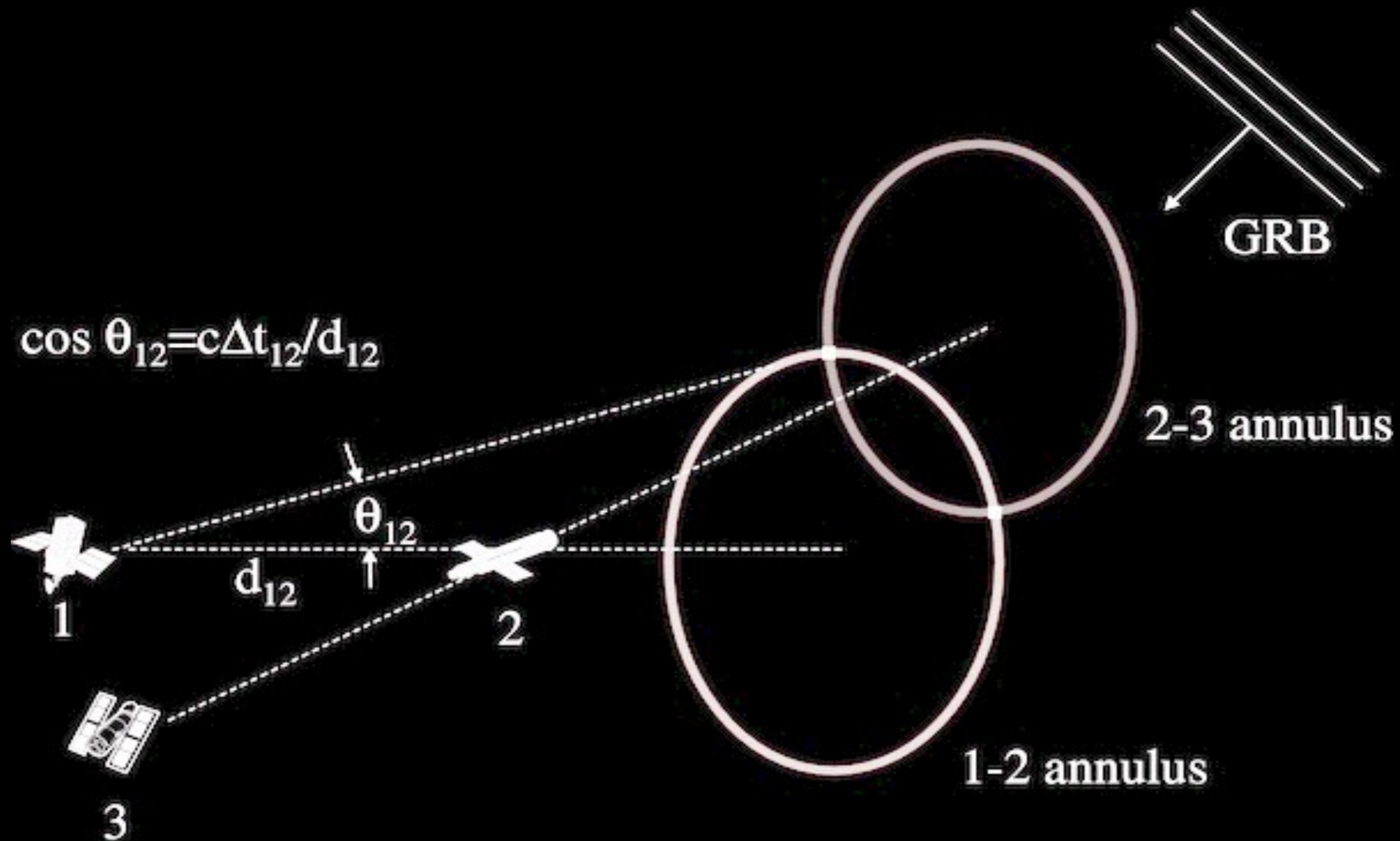
- Accurate positions
- QG tests

## **Limited cost and quick development**

- COTS + in-house components
- Trend in cost reduction of manufacturing and launching QS



# Experiment concept

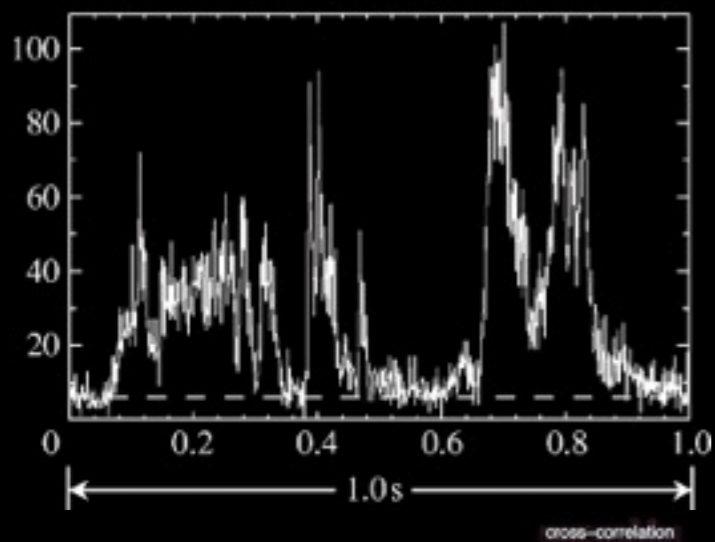




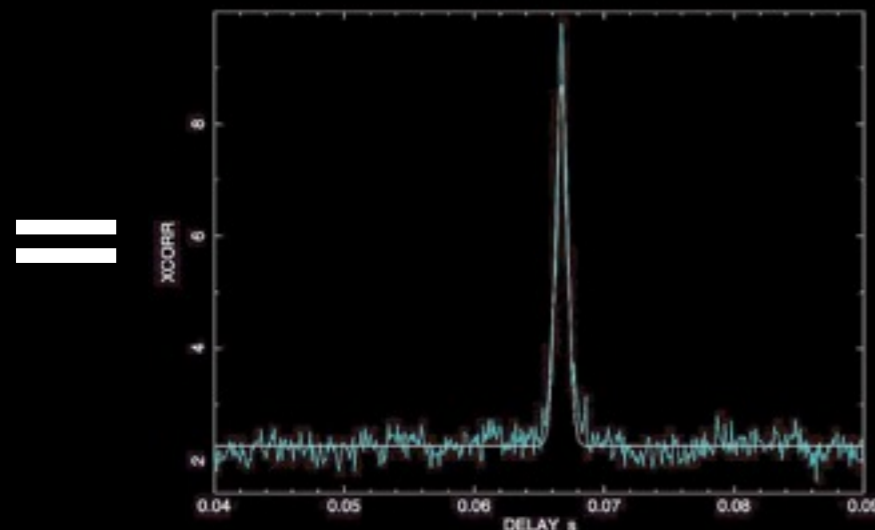
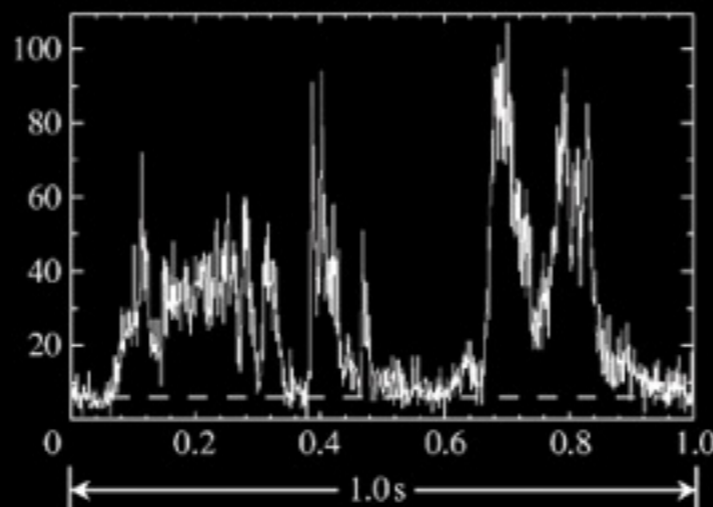
# Experiment concept

1. Measure GRB positions through delays between photons arrival times:

$$\sigma_{\text{Pos}} = (\sigma_{\text{CCF}}^2 + \sigma_{\text{sys}}^2)^{0.5} \times c / \langle B \rangle / (N - 1 - 2)^{0.5}$$



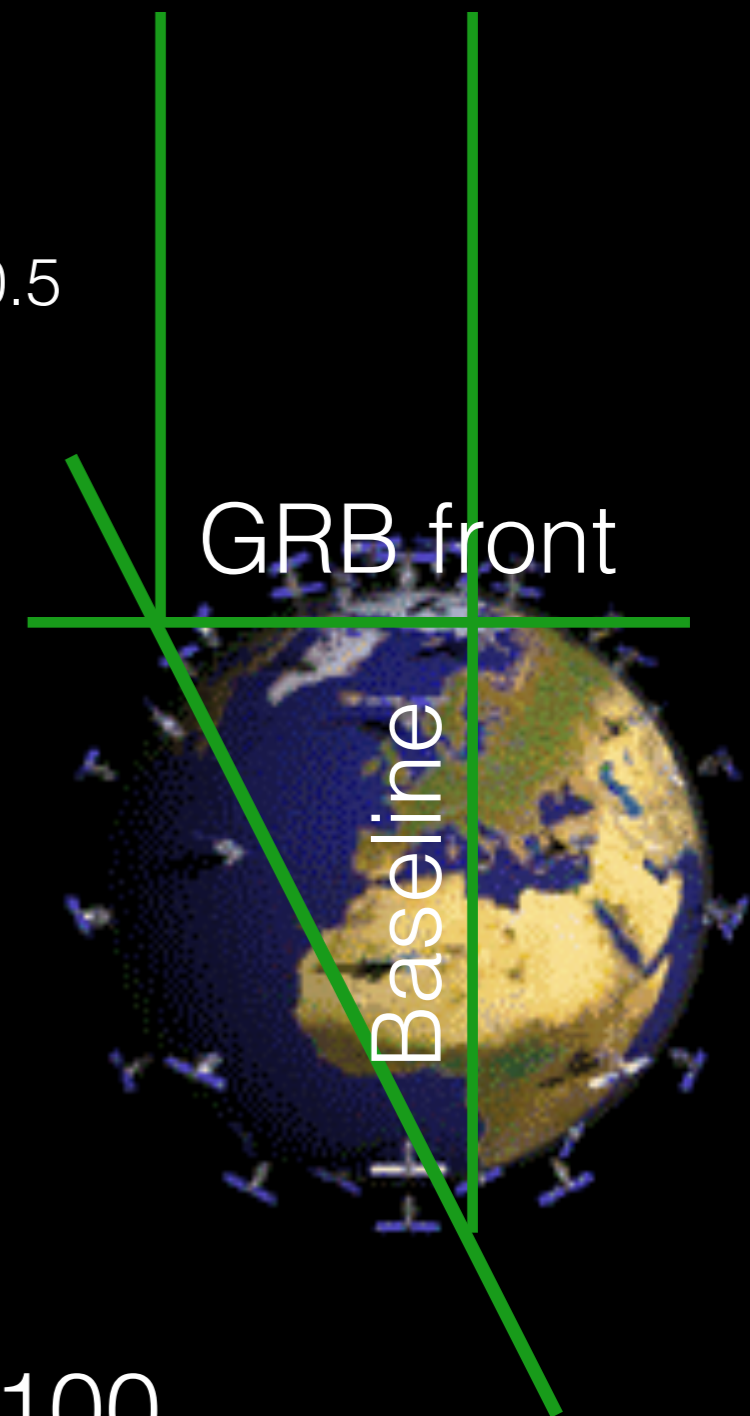
+



$$\sigma_{\text{CCF}} \sim 10 \mu\text{s}$$

$$\sigma_{\text{Pos}} \sim 10 \text{arcsec}$$

$$\text{if } \langle B \rangle \sim 7000 \text{km}, N \sim 100$$

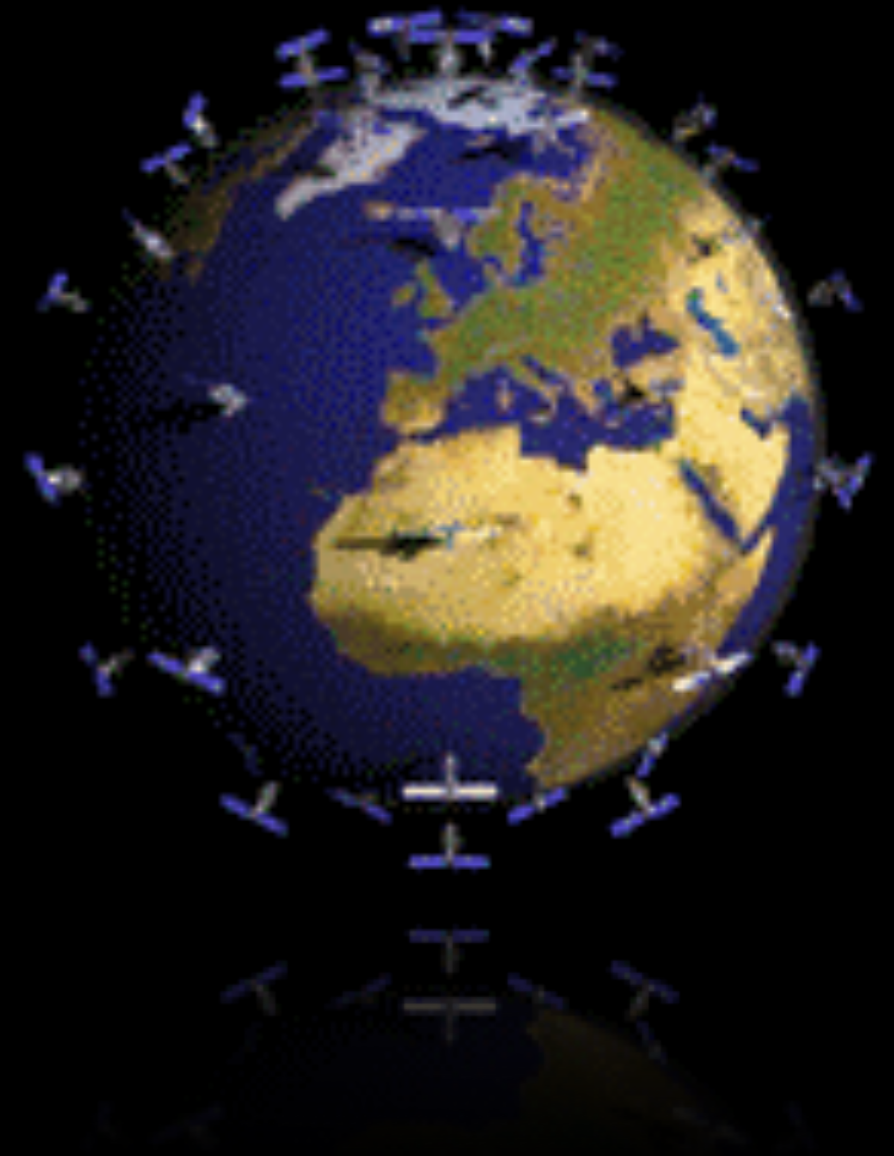


# Experiment concept

2. Add the signal from different units

Total collecting area  $50\text{-}100\text{-cm}^2 \times 100\text{-}200 = 0.5\text{-}2 \text{ m}^2$

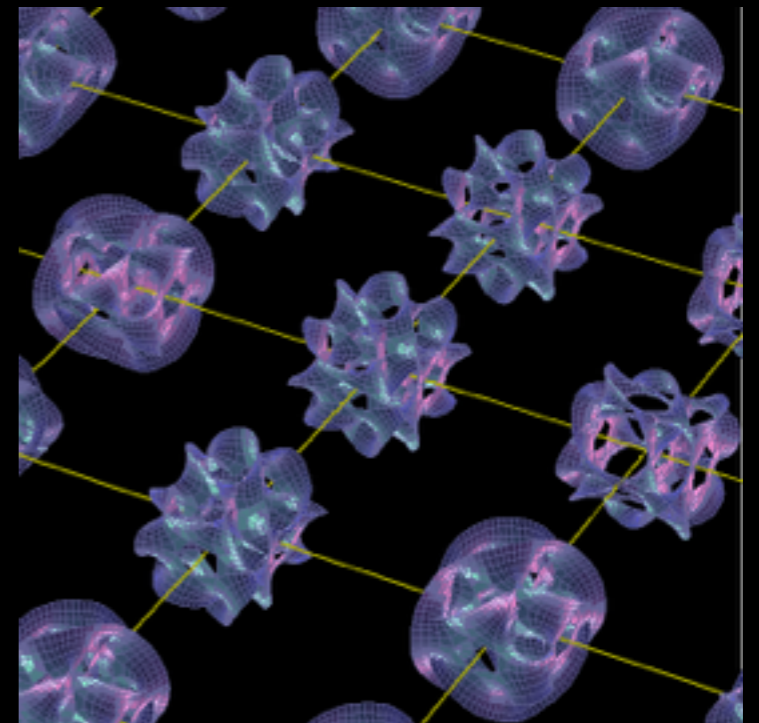
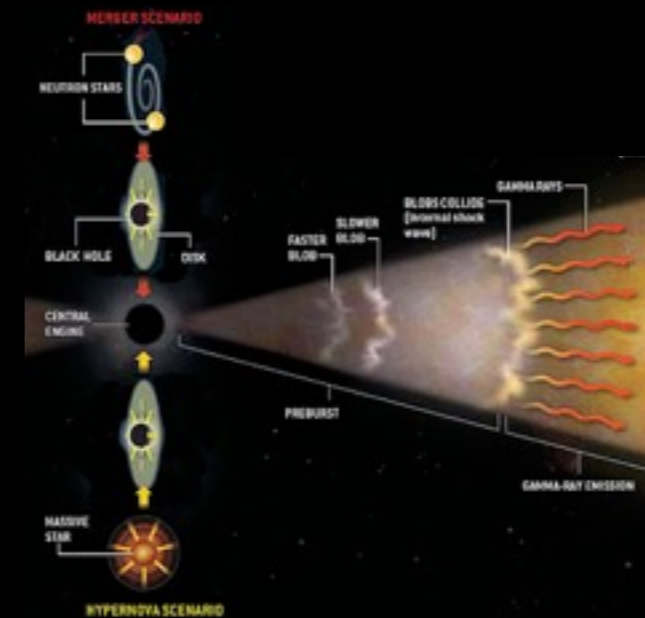
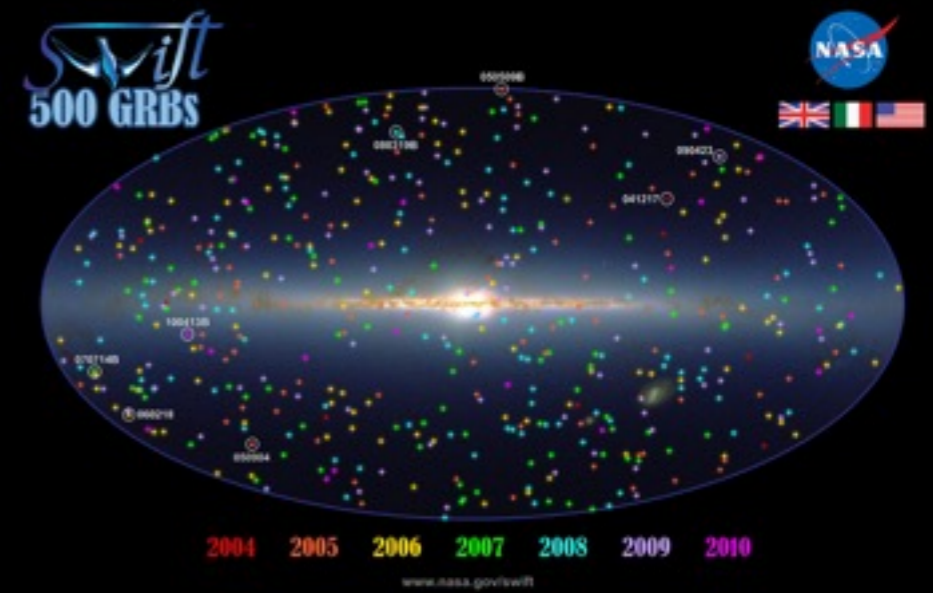
Transient fine (sub $\mu$ s-ms)  
temporal structure



How to *promptly* localise a GRB  
*prompt* event?

How to construct a GRB  
engine?

Which is the ultimate granular  
structure of space-time?



# Requirements

Scientific:

Arcmin-arcsec positions of ~a few dozen GRB/yr

Prompt(minute) localisation

sub- $\mu$ s timing

$\Delta t/\Delta E \sim 3\mu\text{s}/100\text{keV}$   $30\mu\text{s}/1\text{MeV} \longrightarrow M_{\text{QG}} \sim M_{\text{Planck}}$

# Requirements

System:

≈from a few to hundreds detectors

single collecting area  $\geq 50\text{cm}^2$

total collecting area  $\geq 1\text{m}^2$

Energy range 3-10 — 300-1000 keV

Temporal resolution a few hundred ns

Position reconstruction of each satellite  $< 300\text{m}$

Absolute time reconstruction  $< 100\text{ ns}$

Download full burst info in minutes

# Spacecraft

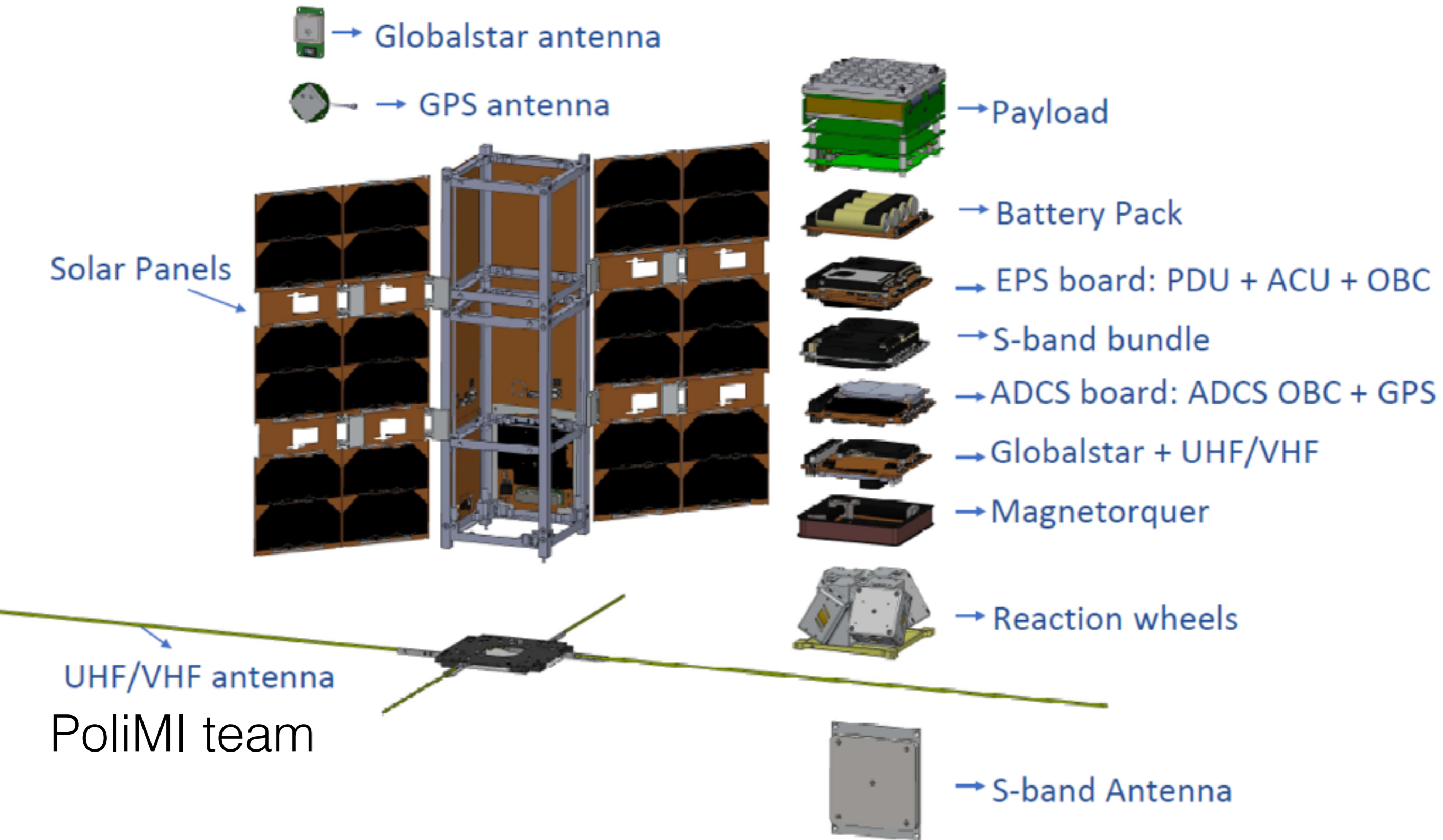
3U minimum, simplest basic configuration

50 cm<sup>2</sup> detector: Pathfinder

6U more performing configuration

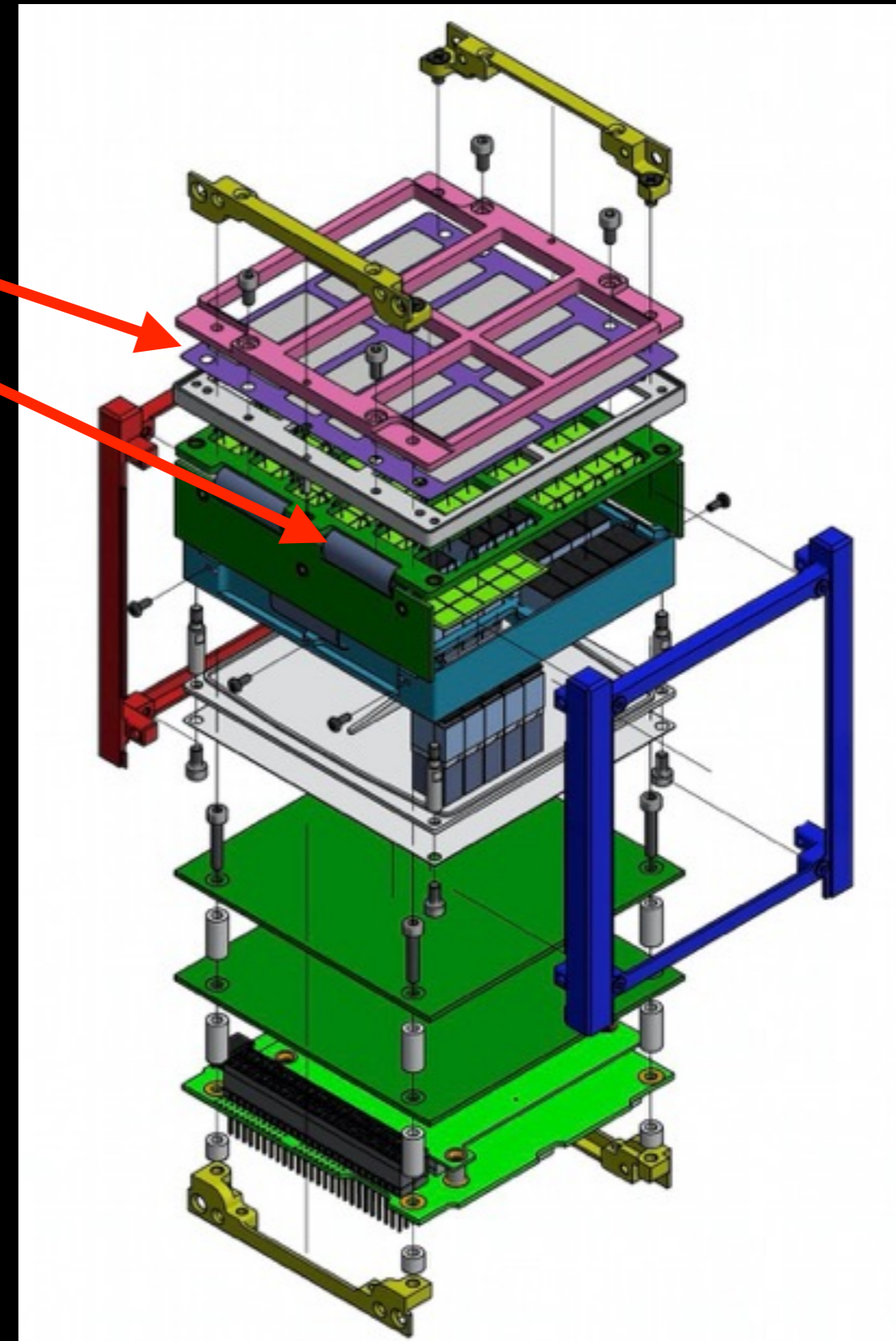
~200cm<sup>2</sup> detector, more accurate GPS, more accurate AOCS: Full Constellation

# Spacecraft



# Payload concept

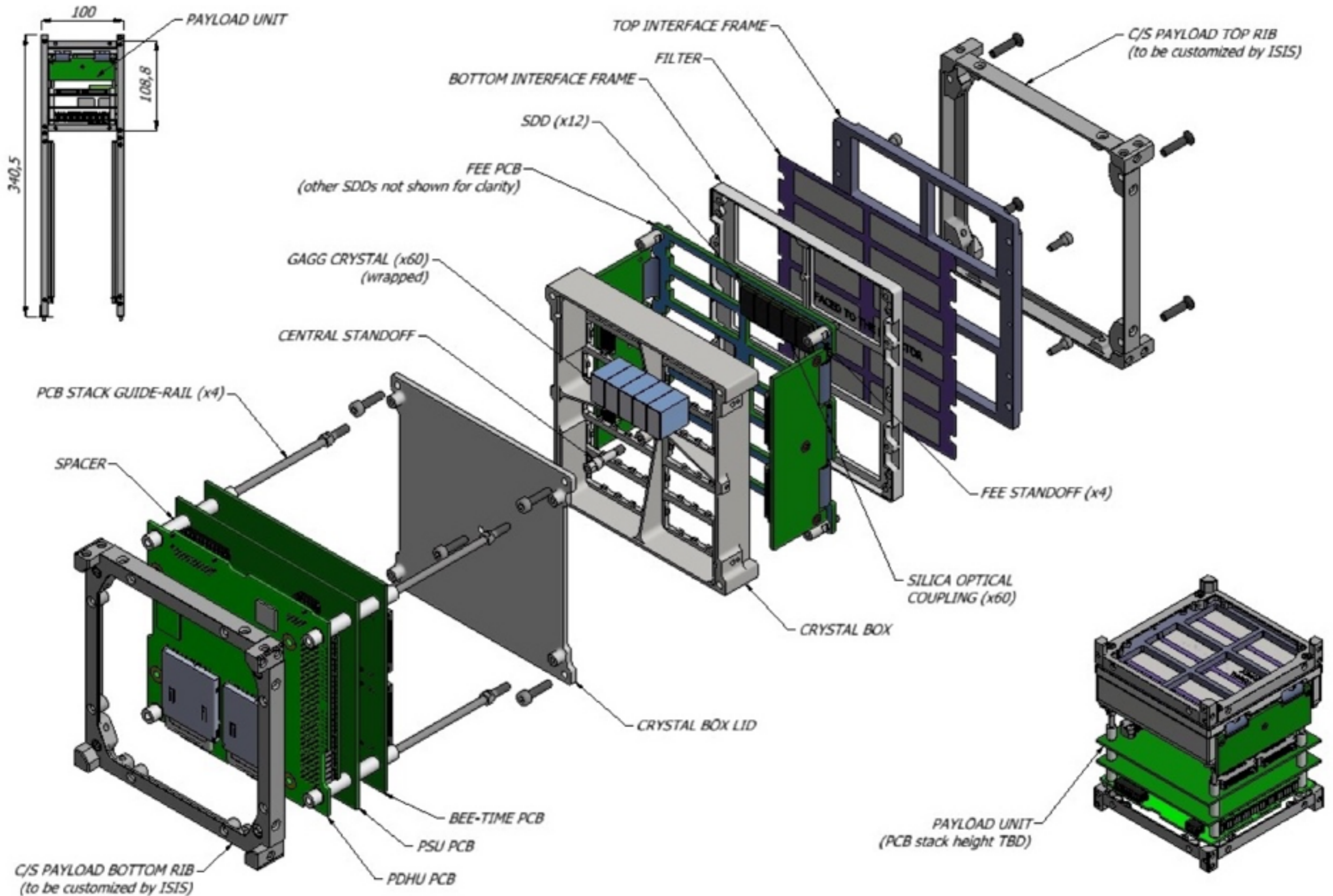
- Photo detector, SDD  
Scintillator crystal GAGG
- 5-300 keV (3-1000 keV)
- $\geq 50 \text{ cm}^2$  coll. area
- a few st FOV
- Temporal res.  $\leq 300 \text{ nsec}$
- $\sim 1.6 \text{ kg}$



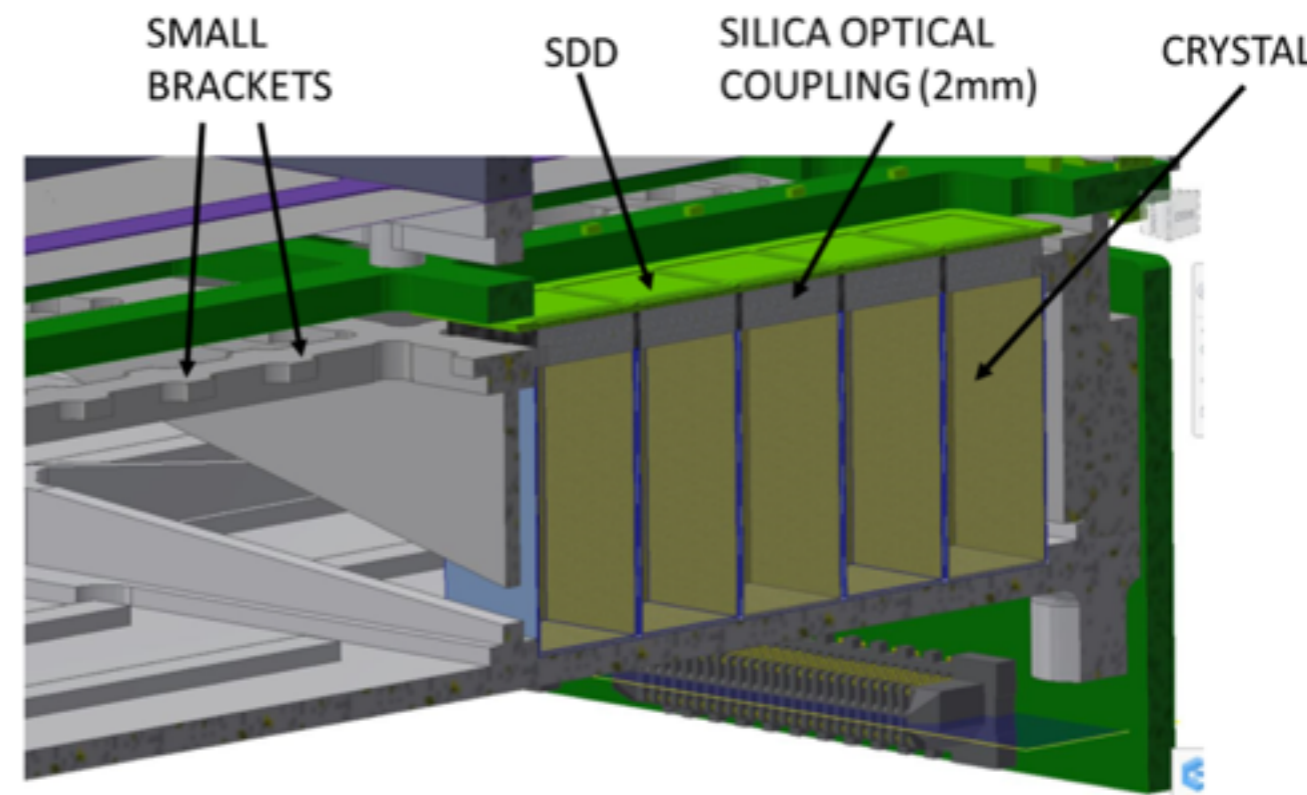
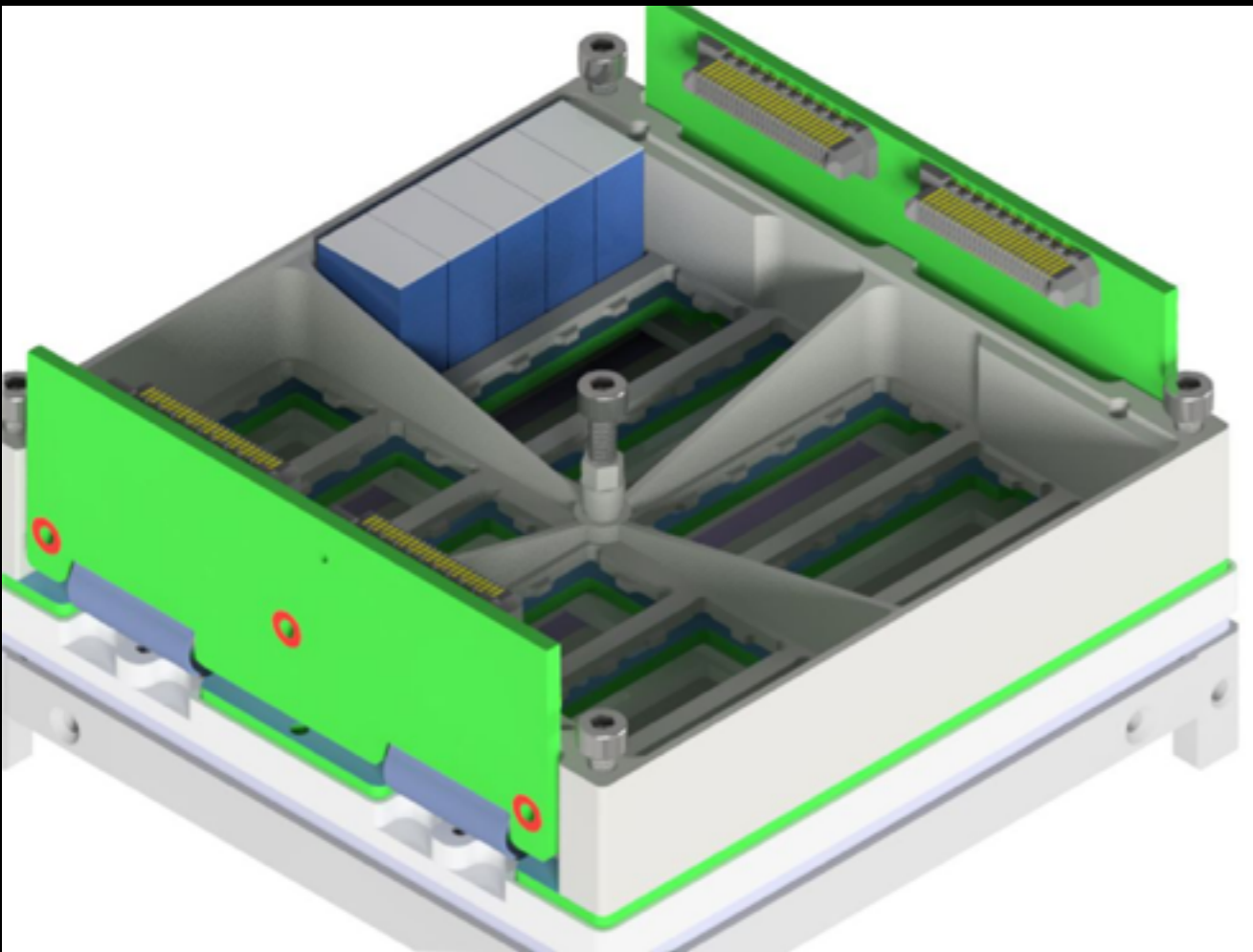
Fuschino+2018, 2020  
Evangelista+2020  
Campana+2020



# Payload design

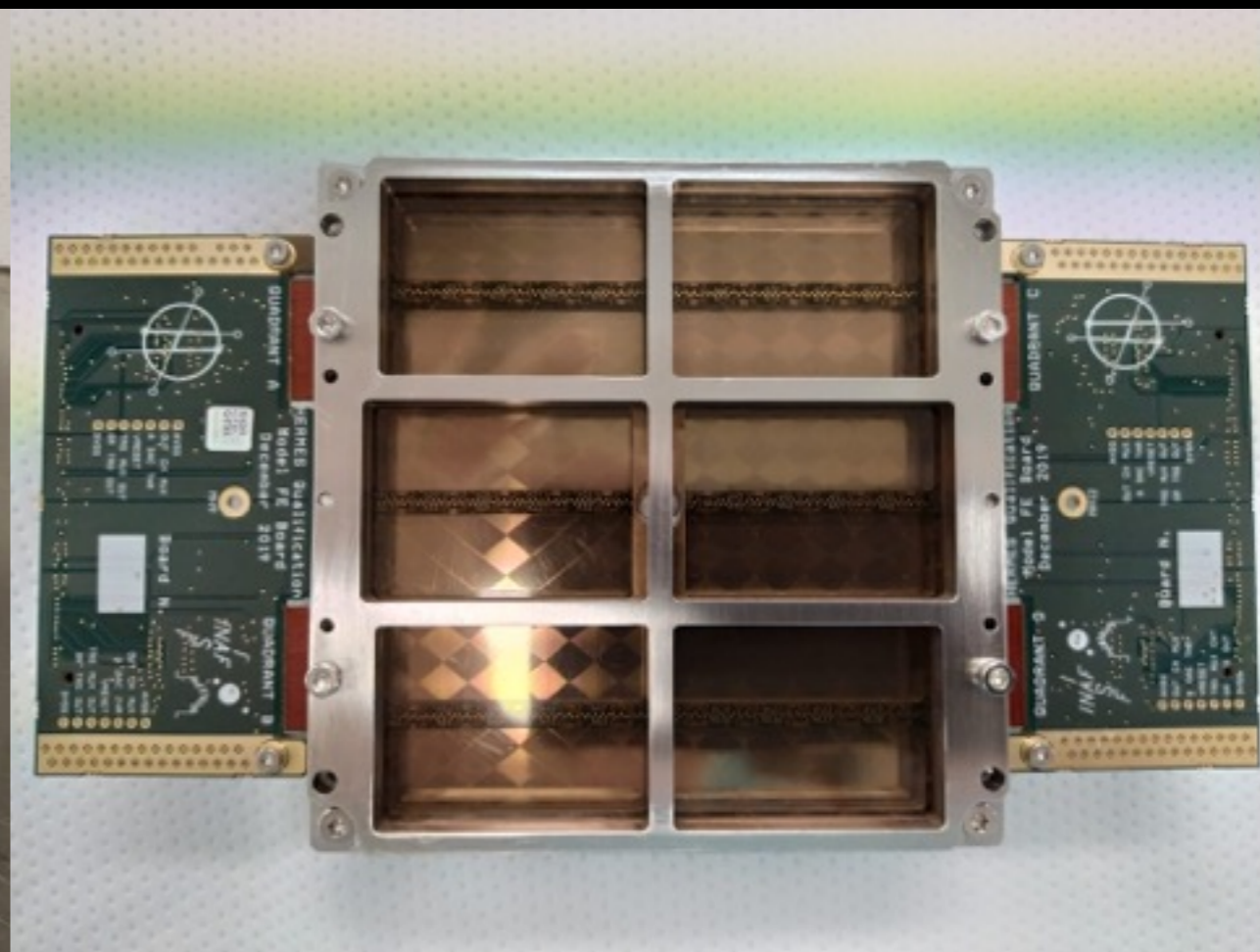
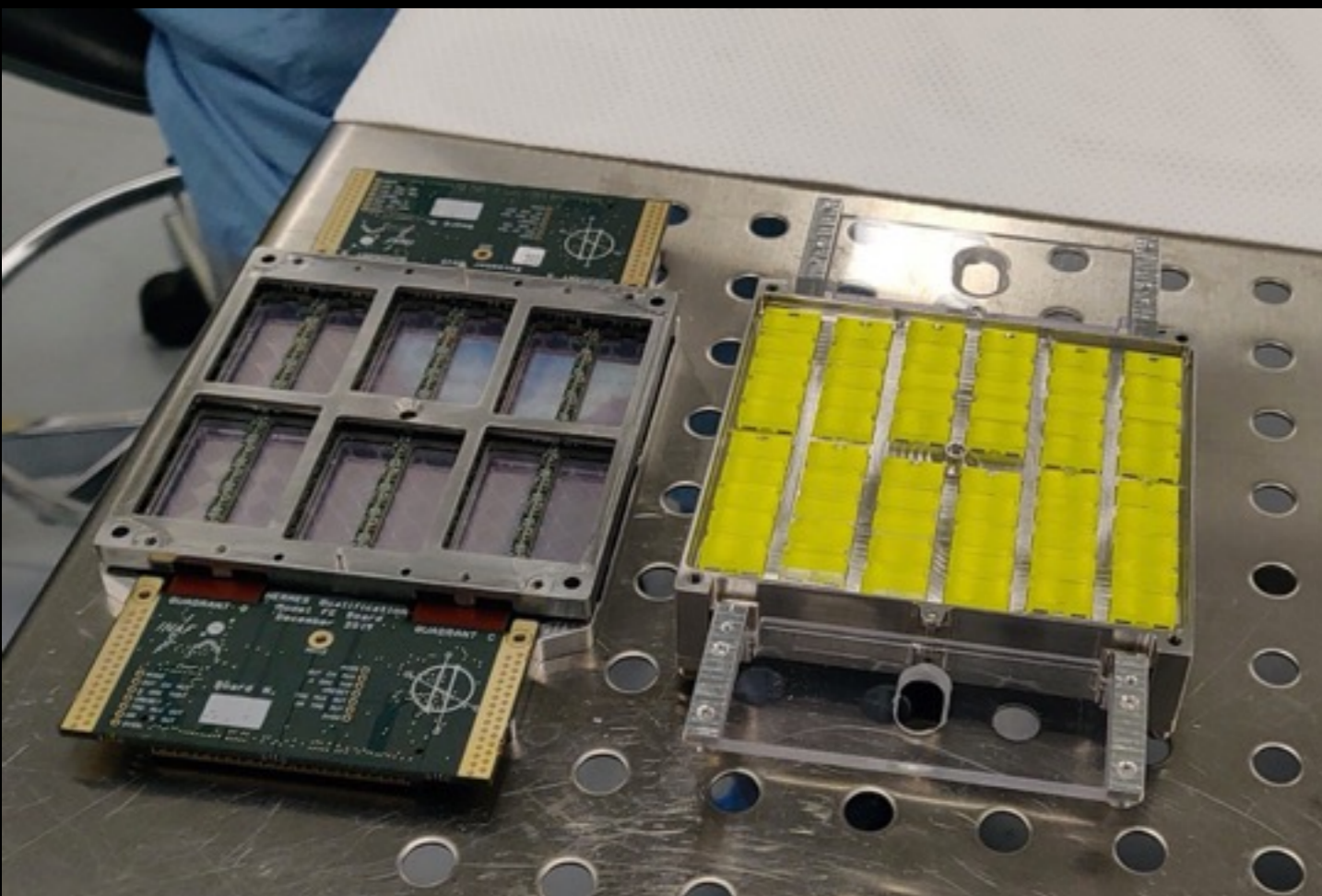


# Detector design



Stainless steel crystal box + tungsten layers on bottom and sides to reduce X-ray background

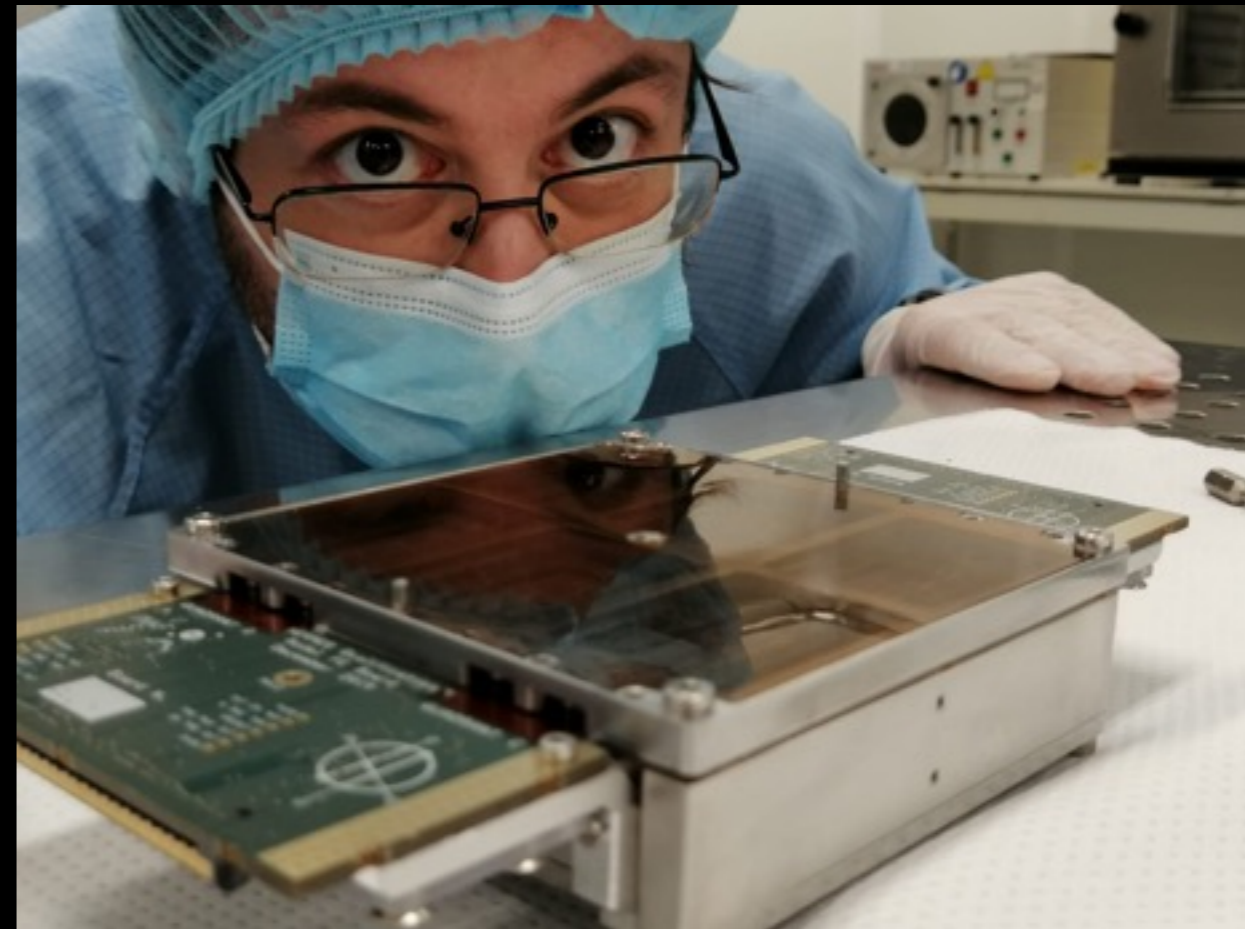
# Hardware



# Payload DM

<http://www.hermes-sp.eu/?p=5010>

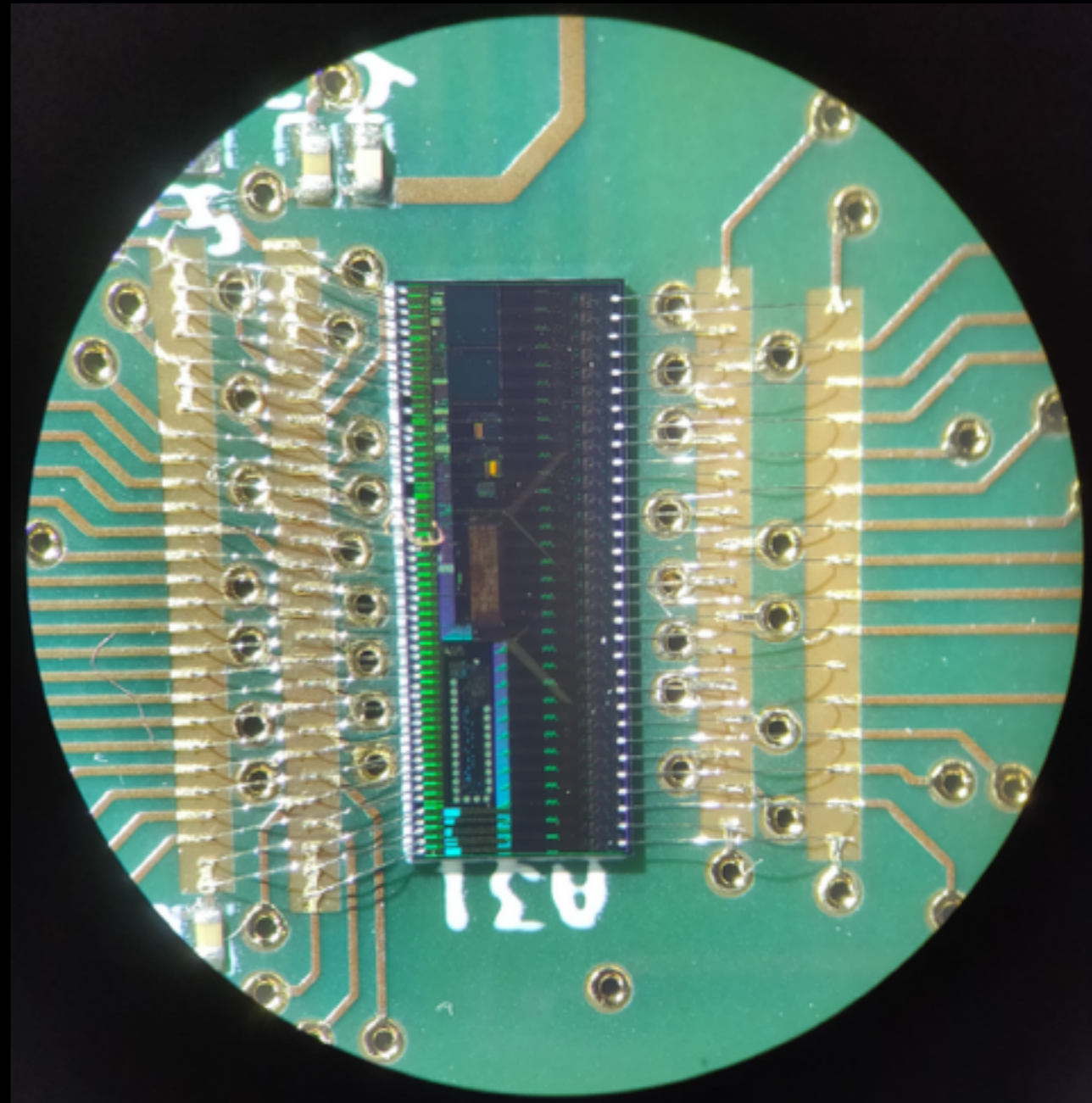
- Assembly, Integration procedure and test plan consolidation
- FEE PCB functional tests
- FEE PCB (preliminary) performances verification
- SDD + ASICs power consumption verification
- Absence of channel-to-channel electrical cross-talk
- Room-temperature performance as expected. Spectroscopic characterisation with  $^{137}\text{CS}$



# Payload DM

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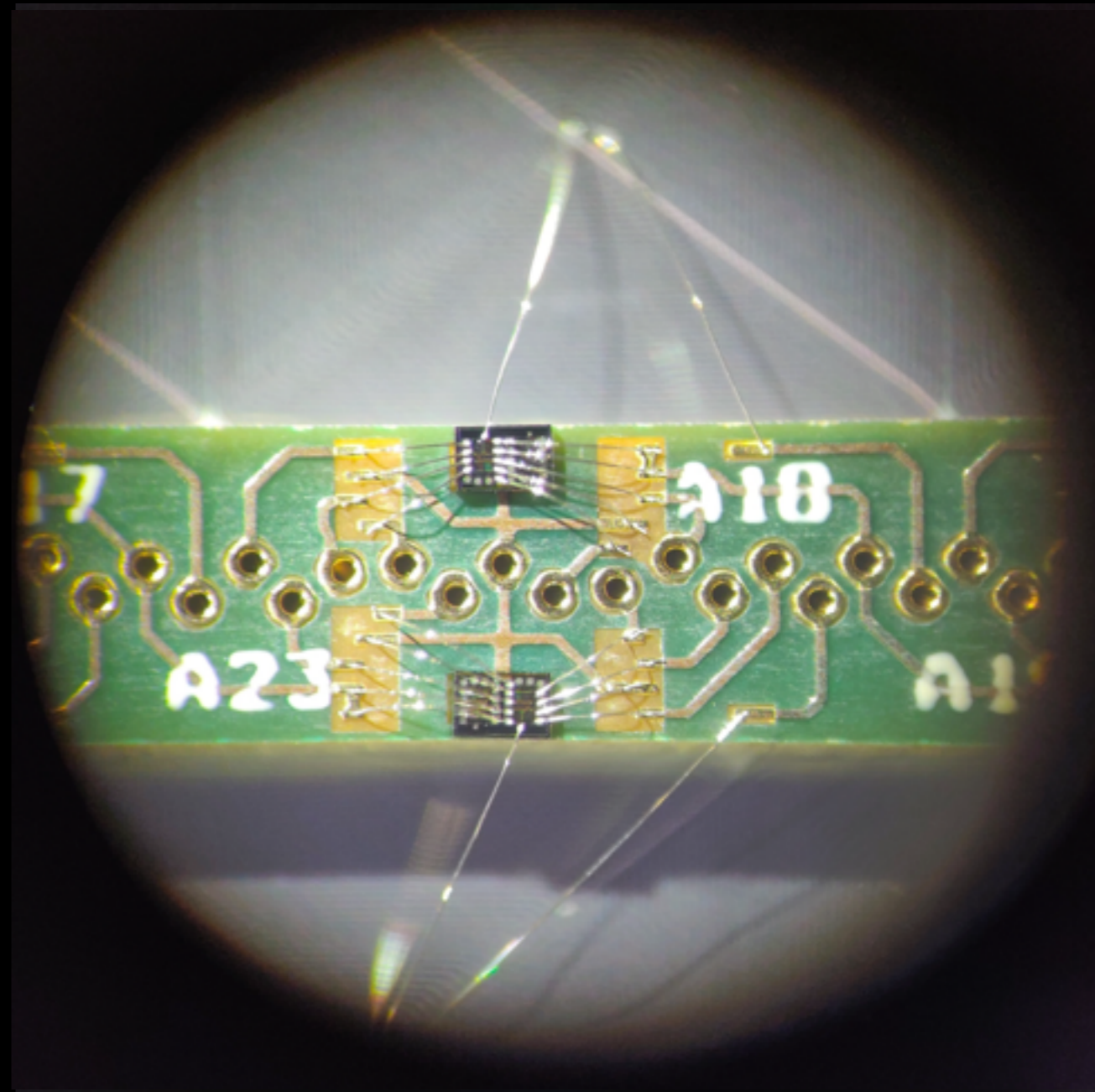
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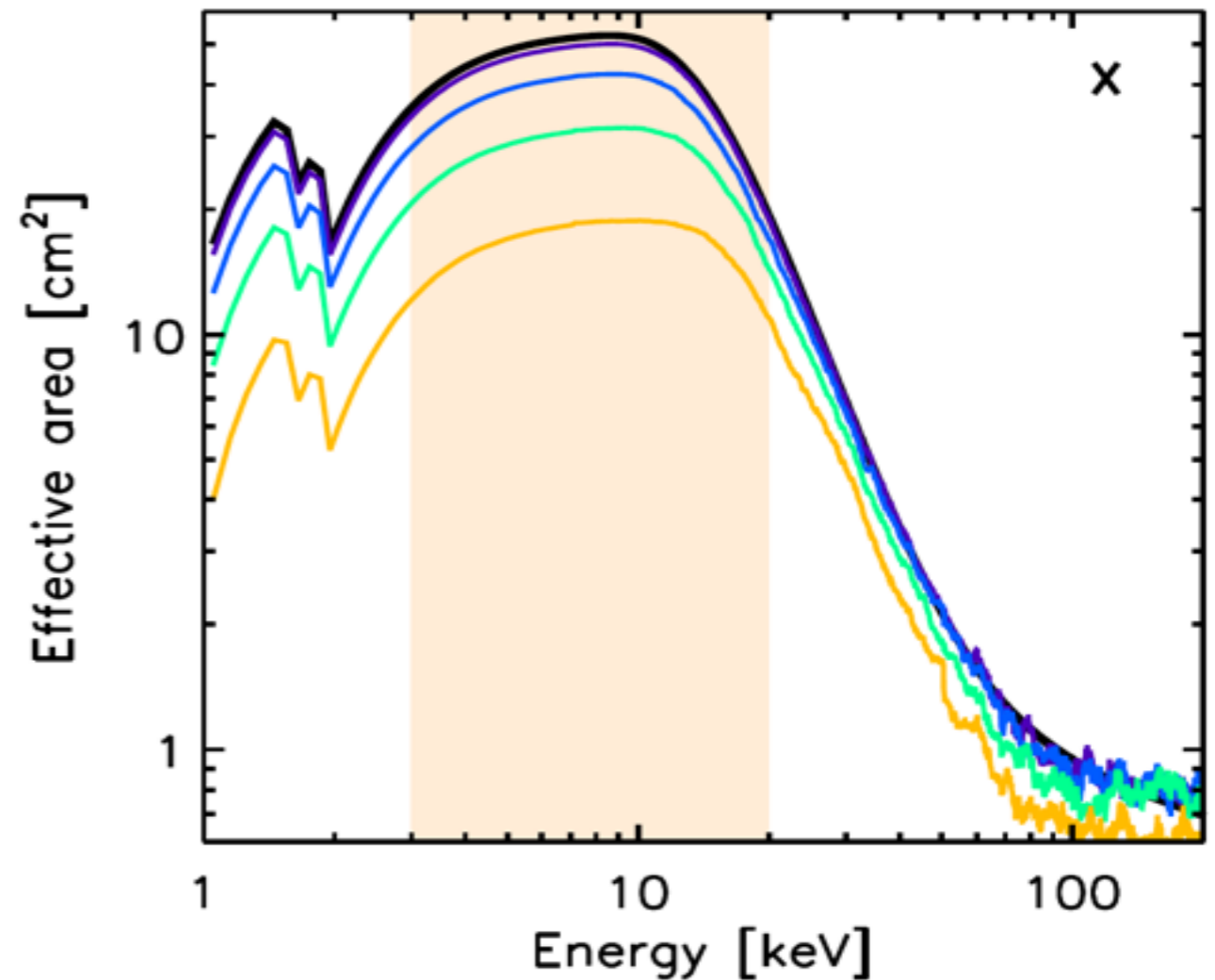
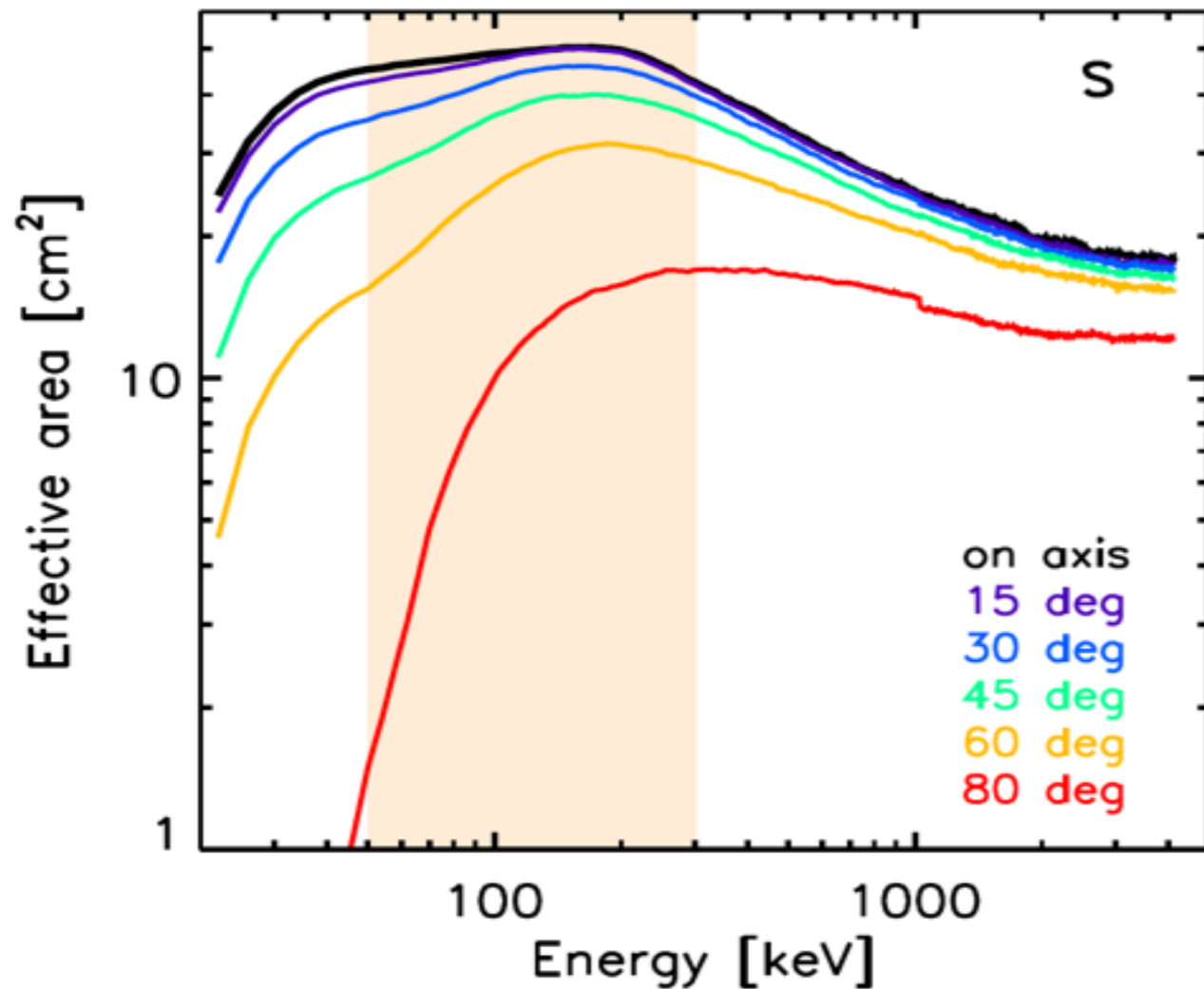
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# HERMES performances

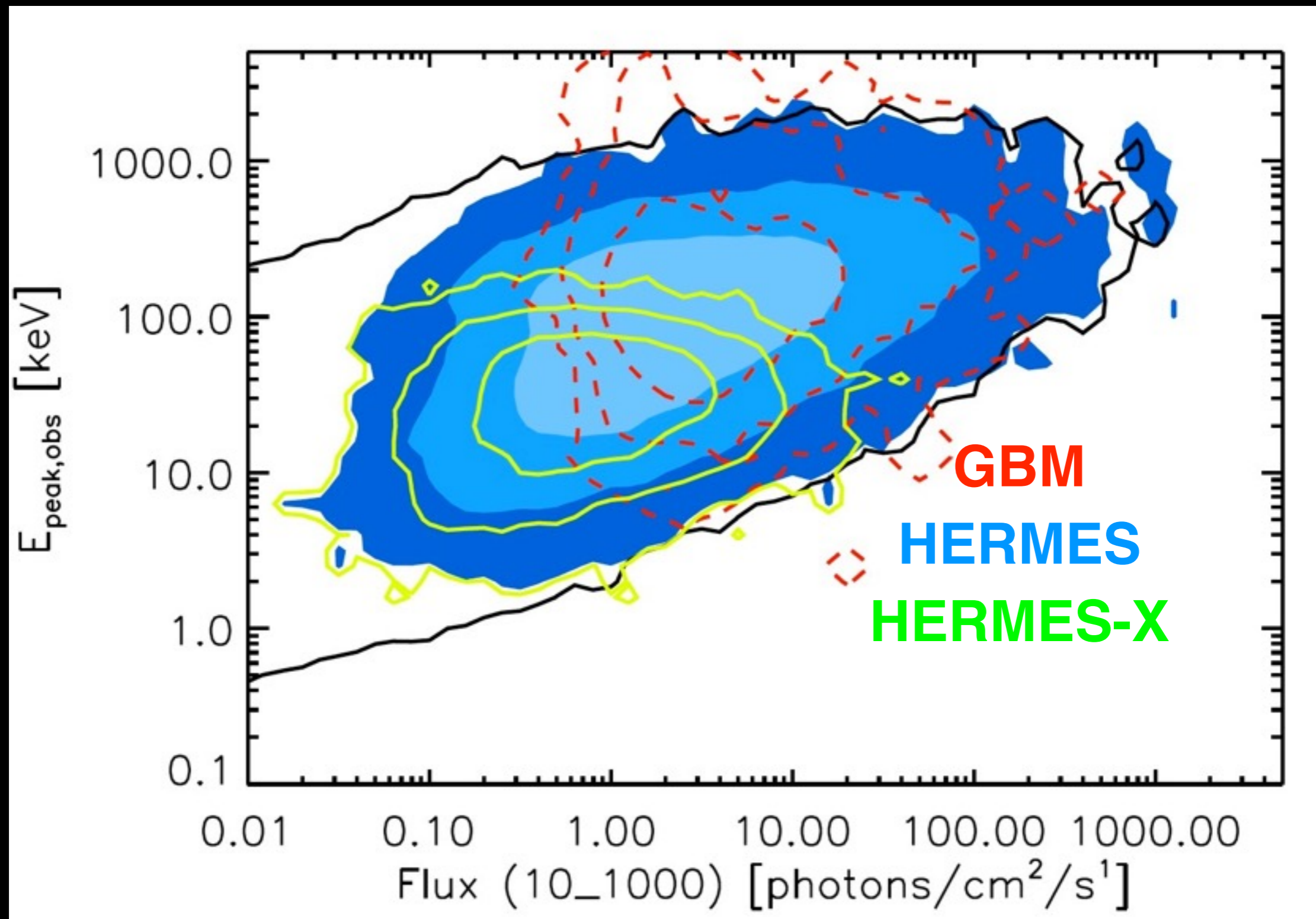


Background: 50-300 keV = 75 counts/s; 3-20 keV 390 counts/s

HERMES vs. GBM: half collecting area but  $\sim 1/3$  lower background and soft energy band

# HERMES performances

Using Ghirlanda/Nava Mock GRB catalog





# HERMES performances

$$\sigma_{\text{Pos}} = 2.4^\circ [(\sigma_{\text{CCF}}^2 + \sigma_{\text{sys}}^2) / (N-3)]^{0.5}$$

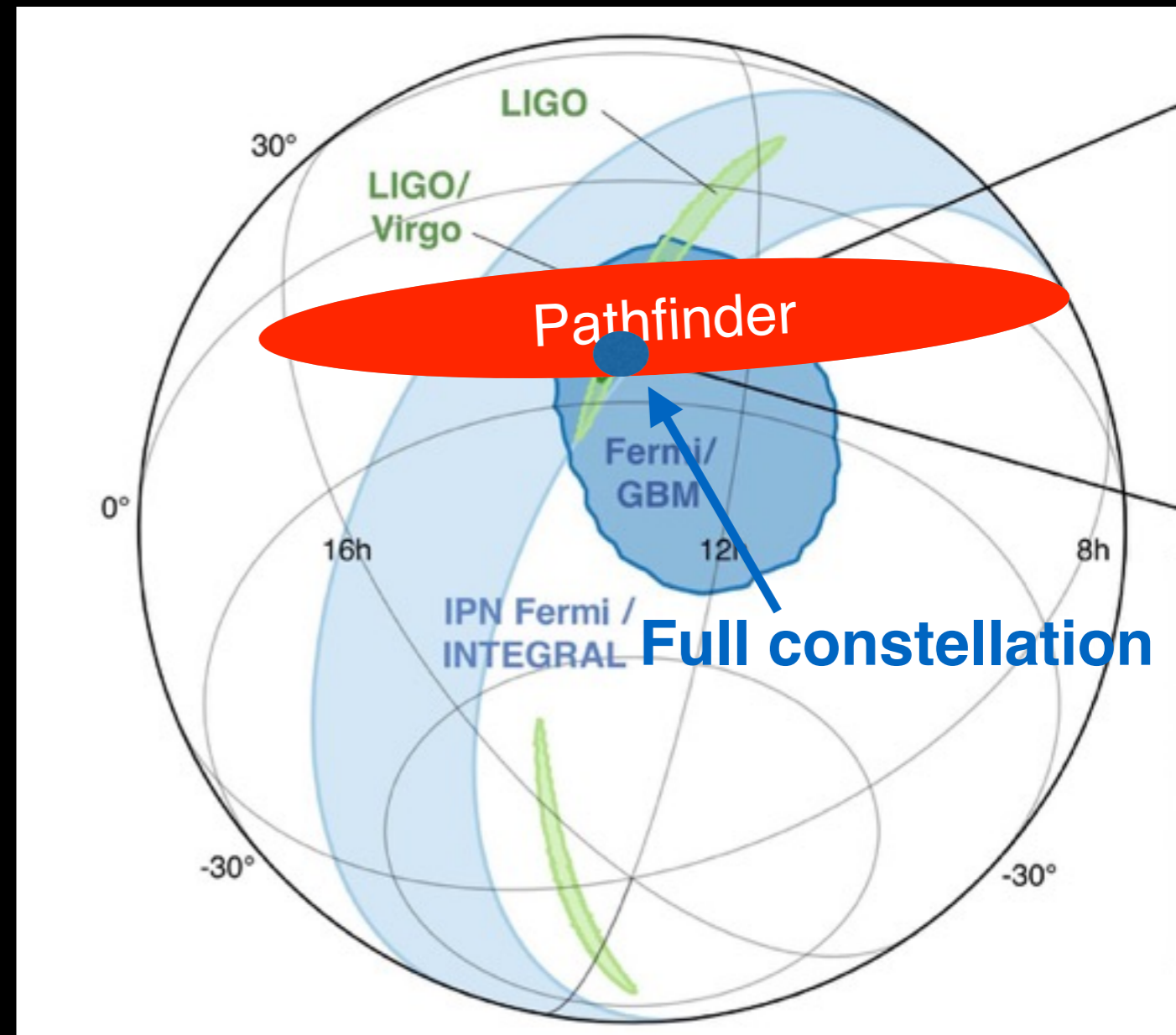
$\langle B \rangle \sim 7000\text{km}$

$N(\text{pathfinder}) \sim 6-8$ , active simultaneously 4-6

$\sigma_{\text{Pos}} \sim 2.4 \text{ deg}$  if  $\sigma_{\text{CCF}}, \sigma_{\text{sys}} \sim 1\text{ms}$

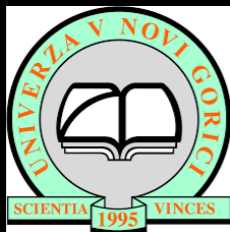
$N(\text{Full constellation}) \sim 100$ , active 50

$\sigma_{\text{Pos}}(\text{FC}) \sim 15 \text{ arcmin}$   
if  $\sigma_{\text{CCF}}, \sigma_{\text{sys}} \sim 1\text{ms}$



# HERMES Institutes

- INAF, ASI, PoliMi, UniCagliari, UniPalermo, UniUdine, UniTrieste, UniPavia, UniFedericoll, UniFerrara, FBK, FPM
- University of Tubingen (Germany)
- University of Eotvos Budapest, C3S (Hungary)
- University of Nova Gorica, Skylabs, AALTA (Slovenia)
- Deimos (Spain)



# Programmatics

Progetto Premiale 2015: **HERMES-Techonogic Pathfinder**

H2020 SPACE-SCI-20: **HERMES-Scientific Pathfinder**

Main objectives:

1. Detect GRBs with simple payload hosted by a 3U CubeSat
2. Study statistical and systematic errors in the CCF determination

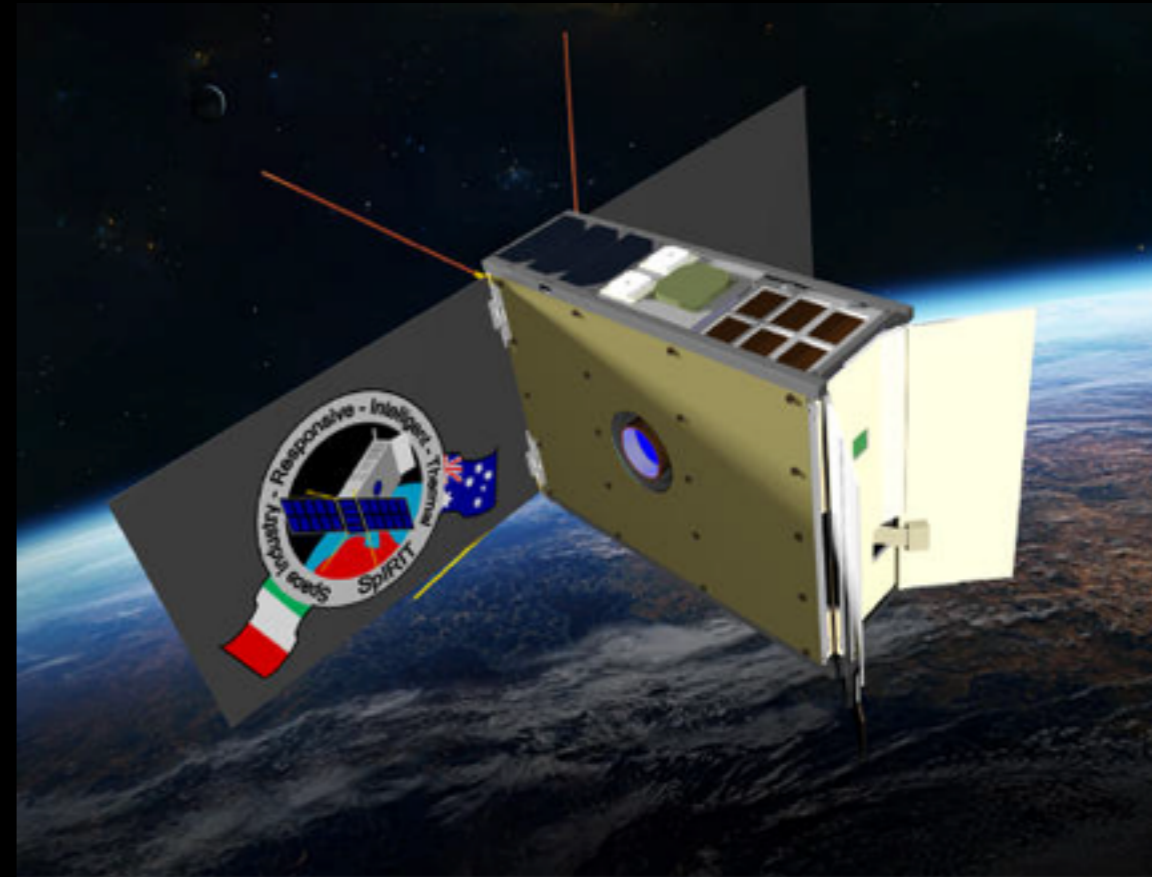
### **3. First GRB localization experiment with $\geq 3$ CubeSat**

- KO May 2018, Nov. 2018
- PDR February-March 2019, DeltaPDR November 2019
- CDR Q3 2020
- QR Q2 2021  $\rightarrow$  PFM1
- AR Q4 2021  $\rightarrow$  FM2+FM3+FM4+FM5+FM6
- Launch 2022, ASI provided

# Next Step

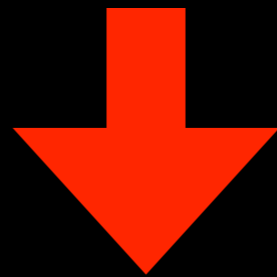
## ◆ Addition of a seventh unit: SpIRIT!

- Australian Space Agency, University of Melbourne
- 6U hosting 1 HERMES payload
- Launch: Q3 2022
- SSO



# Farther Future

- From Pathfinder to full constellation
- From LEO to HEO, Moon, Mars



- sub-arcsec positions
- 1-2 m<sup>2</sup> collecting area

Thanks!